

DUGLAS

134

**MISSILE & SPACE SYSTEMS DIVISION
DOUGLAS AIRCRAFT CO., INC.
ENGINEERING LABORATORIES & SERVICES
TECHNICAL MEMORANDUM**

TO: H. T. Sorensen, A-290
FROM: T. J. Sereno, A-270; ext. 2461
SUBJECT: VENT PORT CHECK VALVE VIBRATION TESTS
COPIES TO: D. D. Hofferth, H. B. Mitchell, W. Keller,
G. Cameron, B. R. Wyatt, Jr., C. Hansen,
D. Dearing, A3-860; T. J. Sereno, A-270;
I. M. Williamson, A-270; C.W. Wilson, A3-860(14)
CATALOG NO. PDL 83607
REPORT NO. TM-DSV4B-ENV-
DATE 8/3/67
REQUESTED BY D. D. Hofferth
E.W.O. 29273 TCD 1T10590
TEST PLAN & ITEM NO. J-20 41412
SALES ORDER 5769-6404
CLASSIFICATION OR RESTRICTION Unclassified
PREPARED BY: Nicolas Der APPROVED BY: T. J. Sereno
N. Der T. J. Sereno
B. J. Shaver Prime Lab. Test Conductor

APPROVAL SIGNATURES CERTIFY THAT ALL REQUIREMENTS OF REPORT HAVE BEEN MET INCLUDING THE REPORTING OF NEW TECHNOLOGY/REPORTABLE ITEMS PER SPB'S 92 AND 93 DISCLOSURES ARE MADE ON FORMS 25-207 AND 25-207-1
NEW TECHNOLOGY: IS CONTAINED IN THIS REPORT, IS NOT CONTAINED IN THIS REPORT

ABSTRACT

INTRODUCTION

Vibration tests were performed for qualification testing of two ~~ASPEB~~ Check Valves. The tests were conducted at the vibration facility of the Douglas Dynamics Laboratory, Santa Monica, California, on June 16 and 17, 1967.

PURPOSE

The purpose of the tests was to determine if the vibration environment would cause degradation in the sealing capability of the valves.

The purposes of this technical memorandum are to describe and document the vibration portions of the tests and to transmit the vibration data obtained from the tests.

N70-75910

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4

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(NASA CR OR TMX OR AD NUMBER)

(CATEGORY)

FACILITY FORM 602



EQUIPMENT

Test Specimen

The test specimen consisted of the following:

Two Vent Port Check Valves, P/N 1B67481-1 A, specimen numbers 10 and 11, Test Plan Item J-20, manufactured by the Circle Seal Company, Anaheim, California.

These components and their associated plumbing and hardware were installed in the vent ports of an Actuation Control Module, Test Plan Item J-3, which is used in the pneumatic power system of the DSV-4B stage. The vibration fixture was a rigid magnesium block, measuring 9 x 9 x 5 inches, secured to the shaker head through an adapter plate. The block, property of the Wyle Laboratories, El Segundo, California, bore the identification W-91640 and weighed 26 pounds.

Test Equipment

ITEM	MANUFACTURER	MODEL	TAG/SERIAL NO.
Oscillograph	CEC	5-124	651909
Galvanometers	CEC	7-326 (4)	*
Accelerometers	Endevco	2226 (3)	*
		2235C	HB 63
Amplifiers	Unholtz-Dickie	8PC	EI-27377
		8PMCVA	EI-11826
Amplifiers	Endevco	2711	HA 28
		2713A (2)	*
Tape Recorders	Ampex	CP-100	611828
		PR-10	0942069
Vibration System	Ling	335	36
PSD Equalizer	Ling	ASDE-80	61
Oscillator	Technical Prod.	TP626	611595-2
Analyzer	Technical Prod.	TP627	611595-3
Power Integrator	Technical Prod.	TP633	611595-1
X-Y Recorder	Moseley	2	963

* Channel assignments and instrumentation information are presented on pages A1 through A3.

Accuracy and repeatability were verified by standard calibration procedures during the periodic calibration and certification of the test equipment.

PROCEDURE

The test assembly was attached to the top of the magnesium block for excitation in the A axis, and to the side of the block for excitation in the B and C axes. Four accelerometers were mounted. The control accelerometer (number 1) was oriented in the direction of the vibration and was used simultaneously for shaker control and data recording. The directions of the axes and locations of the accelerometers are shown on pages B1 through B5.

Tests, consisting of sinusoidal sweeps followed by random excitation exposures, were conducted in each of the three mutually perpendicular axes. The test sequence was completed in each axis before changing to the next axis.

Sinusoidal vibration was applied from 5 to 2000 to 5 cps with the frequency changing at the rate of one octave per minute. Input levels were as follows:

<u>FREQUENCY (cps)</u>	<u>LEVEL</u>
5 - 160	0.06 inch D.A. Disp.
160 - 380	80.0g (0 - peak)
380 - 400	0.012 inch D.A. Disp.
400 - 1200	100.0g (0 - peak)
1200 - 2000	70.0g (0 - peak)

Accelerometer response signals during the sinusoidal testing were recorded on oscilloscope charts.

Random vibration tests were performed in each axis in the following steps:

1. The spectral density equalizer was set to 1/4 power and the desired spectrum was checked visually by examination of the equalizer meters.
2. When the setting followed the spectrum closely, a brief full-power run was made, during which time the signal response of the control accelerometer was recorded on a magnetic tape loop and subsequently analyzed and plotted by a continuous sweep analysis system. The following filter bandwidths were used:

<u>FREQUENCY RANGE (cps)</u>	<u>BANDWIDTH (cps)</u>
20 - 50	5
50 - 100	20
100 - 2000	50

3. When the plot showed a satisfactory spectrum, the regular full-power run was completed. Accelerometer response signals were recorded on 1-inch magnetic tape.

PROCEDURE (continued)

The duration of the random vibration tests was 12 minutes in each axis. Input levels were as follows:

<u>FREQUENCY (cps)</u>	<u>LEVEL</u>
20 - 55	0.2 g ² /cps
55 - 150	+9.0 db/octave
150 - 200	3.5 g ² /cps
200 - 350	-3.0 db/octave
350 - 1000	2.0 g ² /cps
1000 - 2000	-6.0 db/octave

The random vibration magnetic tape data were reduced by the Cascade method with the use of the following filter bandwidths:

<u>FREQUENCY RANGE (cps)</u>	<u>BANDWIDTH (cps)</u>
8 - 16	1.1
16 - 32	3
32 - 65	6.5
65 - 130	13
130 - 260	26
260 - 520	52
520 - 1000	104
1000 - 2000	208

Functional requirements were monitored by Prime Laboratory (ADDO) personnel during the vibration testing.

RESULTS AND DISCUSSION

Sinusoidal data are presented as plots of Acceleration versus Frequency on the following pages:

<u>AXIS</u>	<u>PAGE NUMBER</u>
A	A4 through A7
B	A8 through A11
C	A12 through A15

RESULTS AND DISCUSSION (continued)

Equalization plots of the random vibration inputs, as approved by the Task Force representative, are presented on pages A16 through A18.

Random vibration data are presented as plots of Power Spectral Density versus Frequency on the following pages:

<u>AXIS</u>	<u>PAGE NUMBER</u>
A	A19 through A22
B	A23 through A26
C	A27 through A30

Post-test examination revealed no apparent damage to the test specimens resulting from the vibration exposures.

ATTACHMENTS

Pages A1 through A30

Pages B1 through A5

DOUGLAS

INSTRUMENTATION CHANNEL ASSOC

TEST TITLE VENT PORT CHECK VALVES.O. 5769-6101 E.N.O. 29273 T.C.P. IT 10590P.N. SPECIMEN 18-67482-1 SPEC. # 10, # 11ENGINEER David/Dar EXT. 3522TECHNICIAN Jeronim/Makashima

STANDARD ACCEL. TYPE _____ S.A. _____ CAL DATE _____

TAPE REC. W.

RECORDS SPE

OSCILLOGRAPH

PAPER SPEED

TIMING LINES

LOC. NO.	TAPE REC. CHAN	OSC CHAN	XDCR CABLE NO.	MEASUREMENT LOCATION	RESP. AXIS	XCAR MODEL
1	1	1		Control - Input to J-3	A	2235G
2	2	2		On J-3 Module - Input to J-20	A	2226
3	3	3		On J-3 Module - Input to J-20	B	2226
4	4	4		On J-3 Module - Input to J-20	C	2226
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NOTES & SKETCHES

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①

DT DYNAMIC TEST

PP.300 S/N 611828 FACILITY 33501 UNIT 113
 15 IN SEC 27 Kc TAPE REEL NO _____
 TYPE 5-124 S/N 651909
 1.25 IN/SEC
 SEC PAPER _____

REPORT NO R 6033-1

PAGE A1

SHEET _____

DATE June 17, 1967

LINE ITEM J-20

RUL. NO. Sine & Random

AXIS A

SERIAL NO.	XDCR AMP MODEL	SERIAL NO.	OSC CAL VO MODEL	SERIAL NO.	APE REC AMP MODEL	CALIB. Random	CALIB. Sine UP	CALIB. Sine DOWN	Sens. pc/g_pk	LIN
HB 63	8-IMLVA	BT-11416	7-326	18760		60 g rms	100 g	100 g	25.2	1
NC 63	2713A	KB 25	7-326	13362		300 g rms	100 g	300 g	2.94	2
NC 59	2711	HA 28	7-326	18965	*	100 g rms	100 g	100 g	2.89	3
PB 83	2713A	KB 30	7-326	13379	**	100 g rms	100 g	100 g	2.50	4
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* 5 MIN. AFTER THE BEGINNING WAS CHANGED TO 300 g rms
 ** 5 MIN. AFTER THE BEGINNING WAS CHANGED TO 300 g rms

Fold out 2

DOUGLAS

INSTRUMENTATION CHANNEL ASSIGNMENT

TEST TITLE VENT PORT CHECK VALVES.O. 5769-6404 SWO 29273 TCD IT 10590P.N. SPECIMEN 1B 67481-1 SPEC. # 10, # 11ENGINEER David/Dor EXT 3522TECHNS J. Johnson/Nakashima

STANDARD ACCEL. TYPE _____ S.N. _____ CAL. DUE _____

TAPE REC. NO.

RECORD SPE

OSCILLOGRAPH

PAPER SPEED,

TIMING LINES

LOC. NO.	TAPE REC. CHAN	OSC CHAN	XDCR CABLE NO.	MEASUREMENT LOCATION	RESP. AXIS	XDCR MODEL
1	1	1		Control - Input to J-3	B	2235C
2	2	2		On J-3 Module - Input to J-20	A	2226
3	3	3		On J-3 Module - Input to J-20	B	2226
4	4	4		On J-3 Module - Input to J-20	C	2226
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NOTES & SKETCHES

Foldout #1

FNT DYNAMIC TEST

CR-100 S/N 611828 FACILITY 3354 UNIT 143
 15 IN SEC 27 Kg TAPE REEL NO.
 TYPE 5-124 S/N 651909
 0.25 IN SEC
 SEC PAPER

REPORT NO R 6033 - 1

PAGE A2

SHEET OF

DATE June 16, 1967

LINE ITEM J-20

R.F. NO. Sine & Random

AXIS B

SERIAL NO.	XDCR AMP MODEL	SERIAL NO.	OSC GAL VO MODEL	SERIAL NO.	TAPE REC AMP MODEL	CALIB. Random	CALIB. Sine UP	CALIB. Sine DOWN	Sens. pe/g_pk	LIN
HB 63	8 PG *	EI-27377	7-326	18760		60 g _{rms}	100 g _{per}	100 g _{per}	25.2	1
MC 63	2713A	KB 25	7-326	13362		1000 g _{pk}	100 g _{per}	300 g _{per}	2.94	2
MC 59	2711	HA 28	7-326	18965		1000 g _{pk}	300 g _{per}	300 g _{per}	2.89	3
PB 83	2713A	KB 30	7-326	13379		300 g _{pk}	100 g _{per}	100 g _{per}	2.50	4
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* AMPLIFIER CHANGED TO 8 PHMCVA # EI-11826
 FOR RANDOM RUN

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#8

INSTRUMENTATION CHANNEL ASSOC

TEST TITLE VENT PORT CHECK VALVE
S.O. 5760-610A **ENO** 29273 **TCD** IT 10590
P.N. SPECIMEN 18-67181-1 **SPBC.** # 10, # 11
ENGINEER David/Dor **EXT.** 3522
TECHNS. Jernigan/Nakashima
STANDARD ACCEL. TYPE _____ **S.N.** _____ **CAL DUE** _____

TAPE REC. W
RECORD SF
OSCILLOGRAPH
PAPER SPEED
LIVING LINE

LOC. NO.	TAPE REC. CHAN	OSC CHAN	XDCR CABLE NO.	MEASUREMENT LOCATION	RESP. AXIS	XDCR MODEL
1	1	1		Control - Input to J-3	C	22350
2	2	2		On J-3 Module - Input to J-20	A	2226
3	3	3		On J-3 Module - Input to J-20	B	2226
4	4	4		On J-3 Module - Input to J-20	C	2226
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NOTES & SKETCHES

Folobut #1

DYNAMIC TEST

FR 100 S/N 611828 FACILITY 3357 UNIT 143
 15 IN SEC 27 Kc TAPE REEL NO _____
 TYPE 5-126 S/N 651909
 25 IN SEC
 SEC PAPER _____

REPORT NO R 6033 - 1

PAGE A 3

SHELF OF

DATE June 17, 1967

LINE ITEM J-20

RUN NO. Sine & Random

AXIS C

SERIAL NO.	XDCR AMP MODEL	SERIAL NO.	OSC GALVO MODEL	SERIAL NO.	TAPE REC AMP MODEL	CALIB. Random	CALIB UP	CALIB SINE	CALIB DOWN	Sens. pe/g pk	
PC 63	8 PMCVA	RI-11316	7-326	18760		60 g rms	100 g rms	100 g rms	100 g rms	25.1	1
PC 63	2713A	KB 25	7-326	13362		1000 g rms	1000 g rms	300/1000 g rms	300/1000 g rms	2.96	2
PC 59	2711	KA 28	7-326	18963		100 g rms	30 g rms	100 g rms	30 g rms	2.89	3
PC 83	2713A	KB 30	7-326	13379		1000 g rms	300 g rms	1000 g rms	1000 g rms	2.50	4
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DOUGLAS AIRCRAFT COMPANY + INC.

PAGE NO.
REPORT NO. 6033--1

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION ---

NOTE... SEE PAGE B1
FOR PICK UP LOCATION

COMMENT--- REFERENCE CHANNEL

LEGEND...

UPSWEEP -----

DOWNSWEEP -----

TEST CONDITIONS...

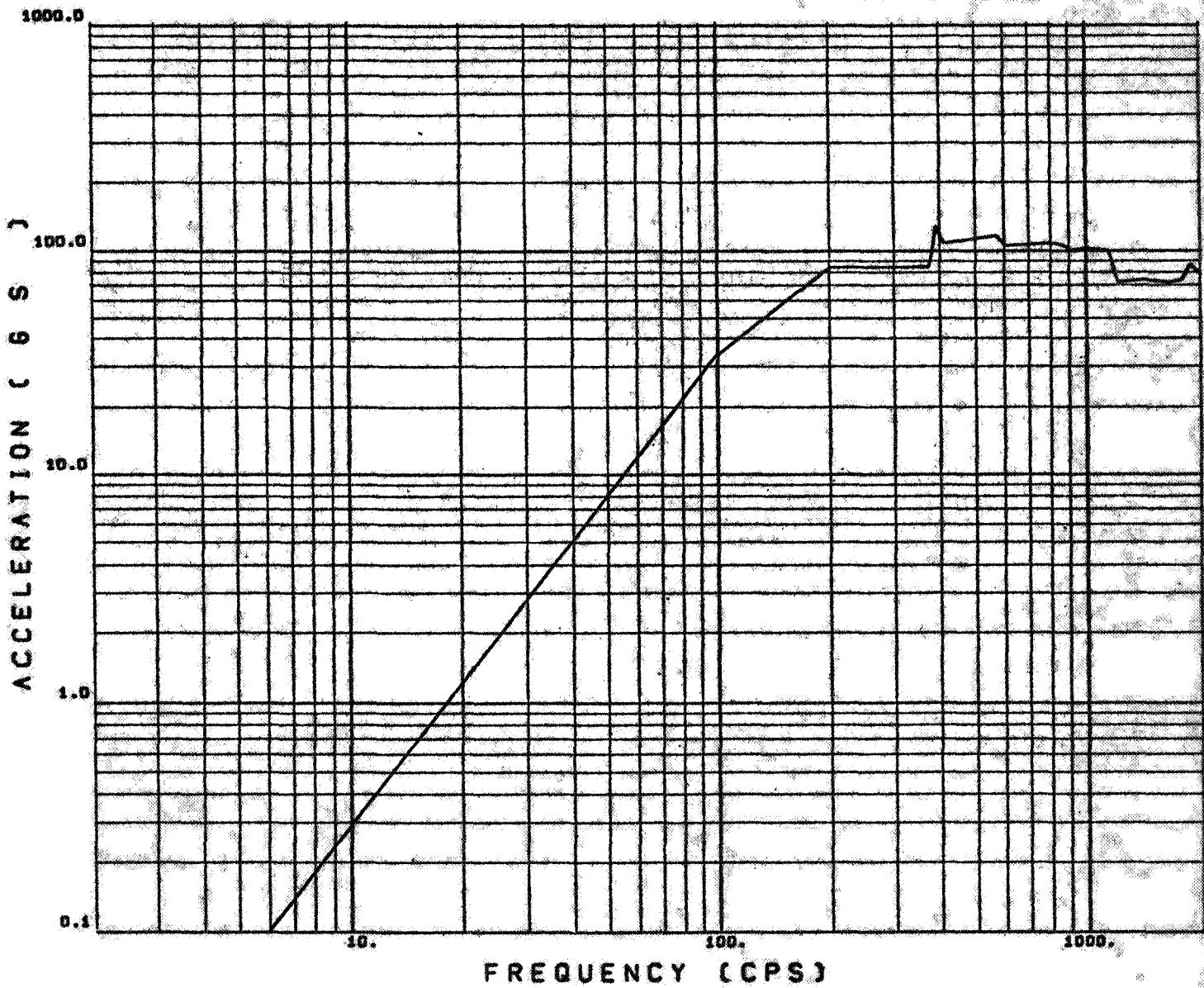
TEST DATE..... 6/17/67

AXIS OF EXCITATION.... A

PICK UP NUMBER (1)... 1 HB65

PICK UP RESPONSE..... A

INPUT ACCEL. PER PAGE..



DOUGLAS AIRCRAFT COMPANY, INC.

PAGE NO. A5
REPORT NO. 6000-1

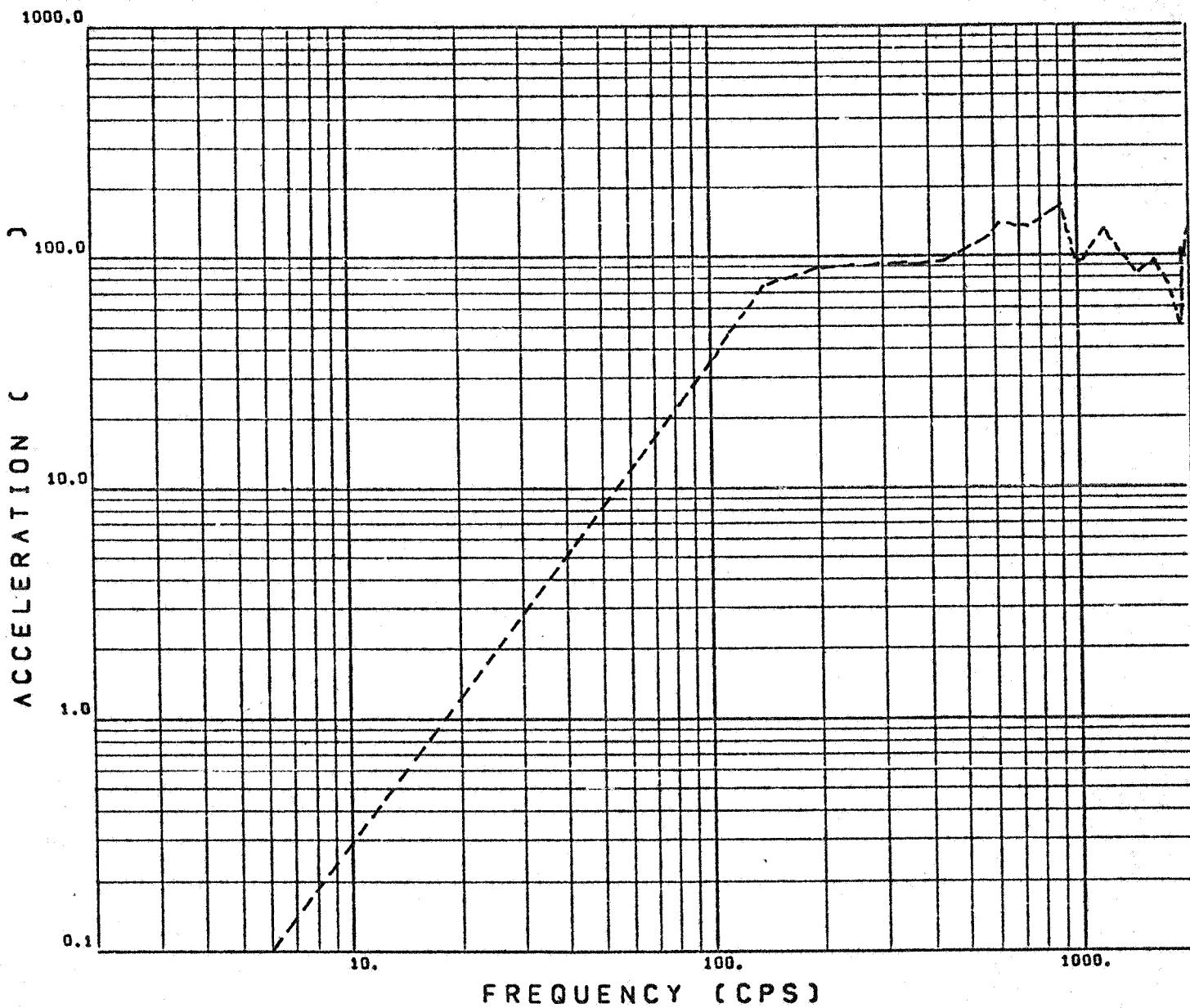
SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION ---
NOTE... SEE PAGE B1
FOR PICK UP LOCATION

LEGEND...
UPSWEEP ———
DOWNSWEEP -----

TEST CONDITIONS....

TEST DATE..... 6/17/67
AXIS OF EXCITATION... A
PICK UP NUMBER (8)... 2 NC63
PICK UP RESPONSE..... A
INPUT ACCEL. PER PAGE.. A4



DOUGLAS AIRCRAFT COMPANY, INC.

A6

PAGE NO.
REPORT NO.

60038-1

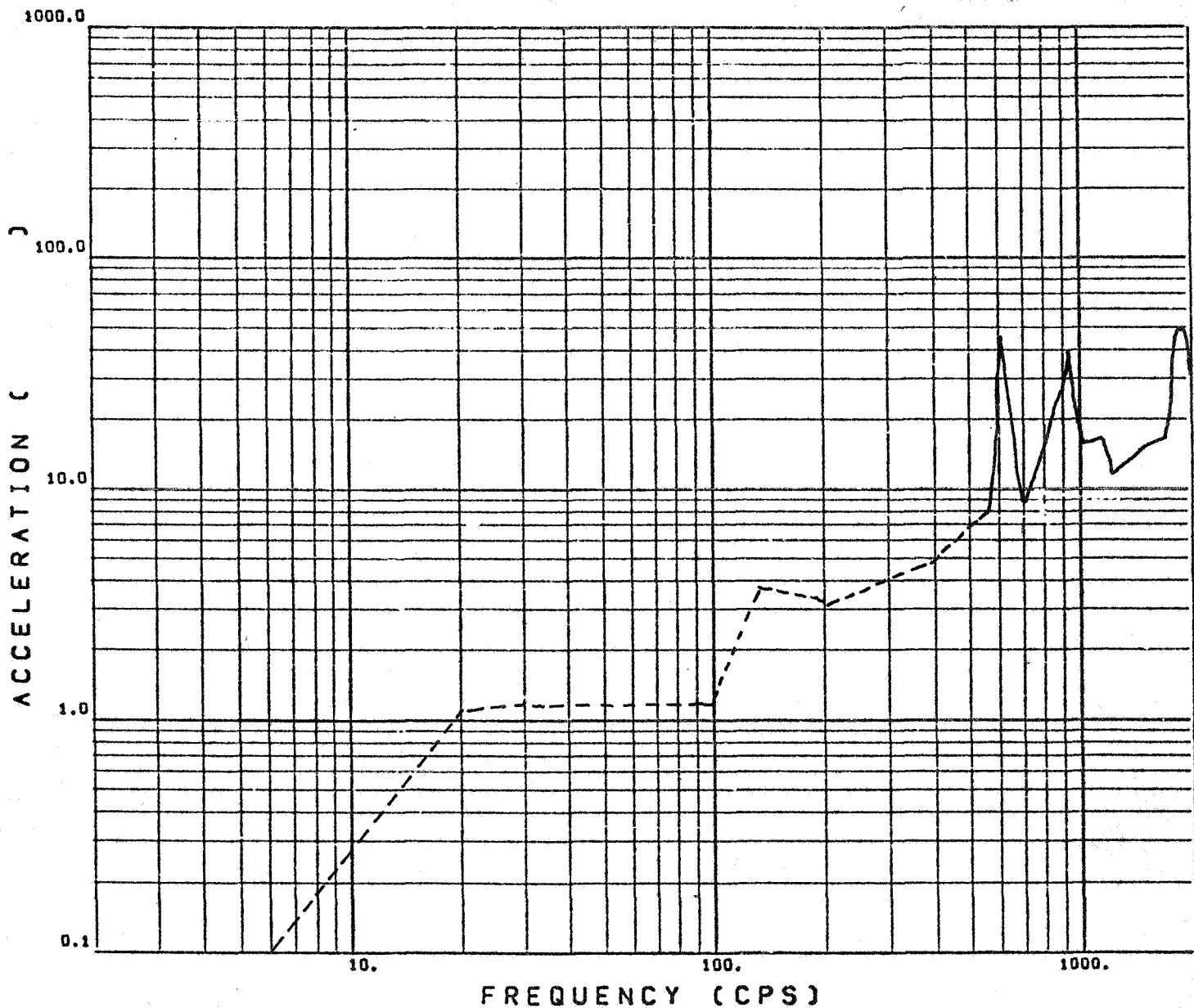
SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION ---
NOTE... SEE PAGE B1
FOR PICK UP LOCATION

TEST CONDITIONS...

TEST DATE..... 6/17/67
AXIS OF EXCITATION.... A
PICK UP NUMBER (6) ... 3 MCSA
PICK UP RESPONSE..... B
INPUT ACCEL.PER PAGE.. A4

LEGEND...
UPSWEEP -----
DOWNSWEEP -----



DOUGLAS AIRCRAFT COMPANY, INC.

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

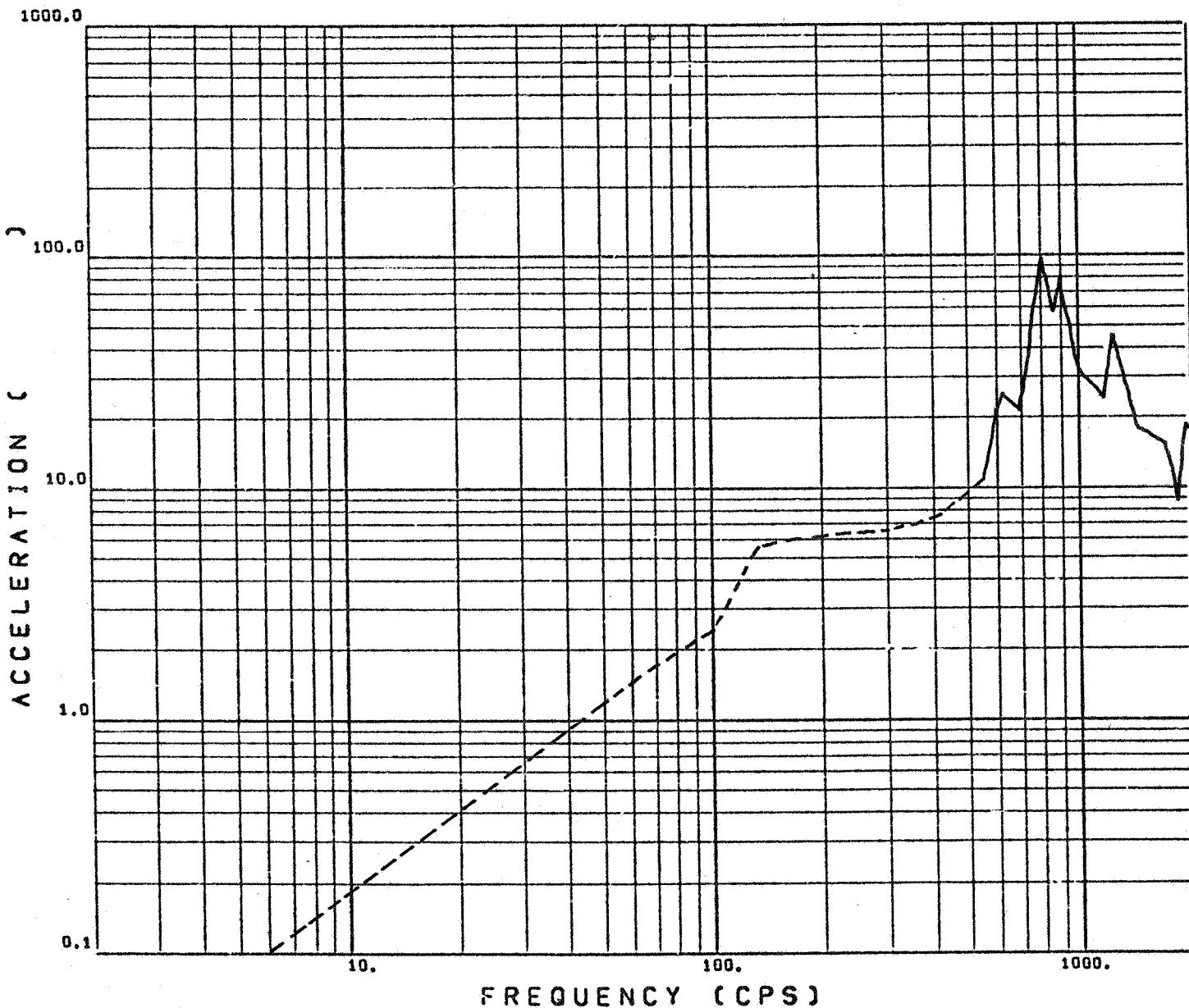
A7
PAGE NO. 8
REPORT NO. R-100R

CONFIGURATION ---
NOTE... SEE PAGE B1
FOR PICK UP LOCATION

LEGEND...
UPSWEEP —
DOWNSWEEP ----

TEST CONDITIONS....

TEST DATE..... 6/17/67
AXIS OF EXCITATION.... A
PICK UP NUMBER (11)... 4 PB33
PICK UP RESPONSE..... C
INPUT ACCEL.PER PAGE.. A4



DOUGLAS AIRCRAFT COMPANY, INC.

A8
PAGE NO.
REPORT NO. R6033-1

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION ---

NOTE... SEE PAGE B3

FOR PICK UP LOCATION

COMMENT---REFERENCE CHANNEL

LEGEND...

UPSWEET -----

DOWNSWEEP -----

TEST CONDITIONS....

TEST DATE..... 6/16/67

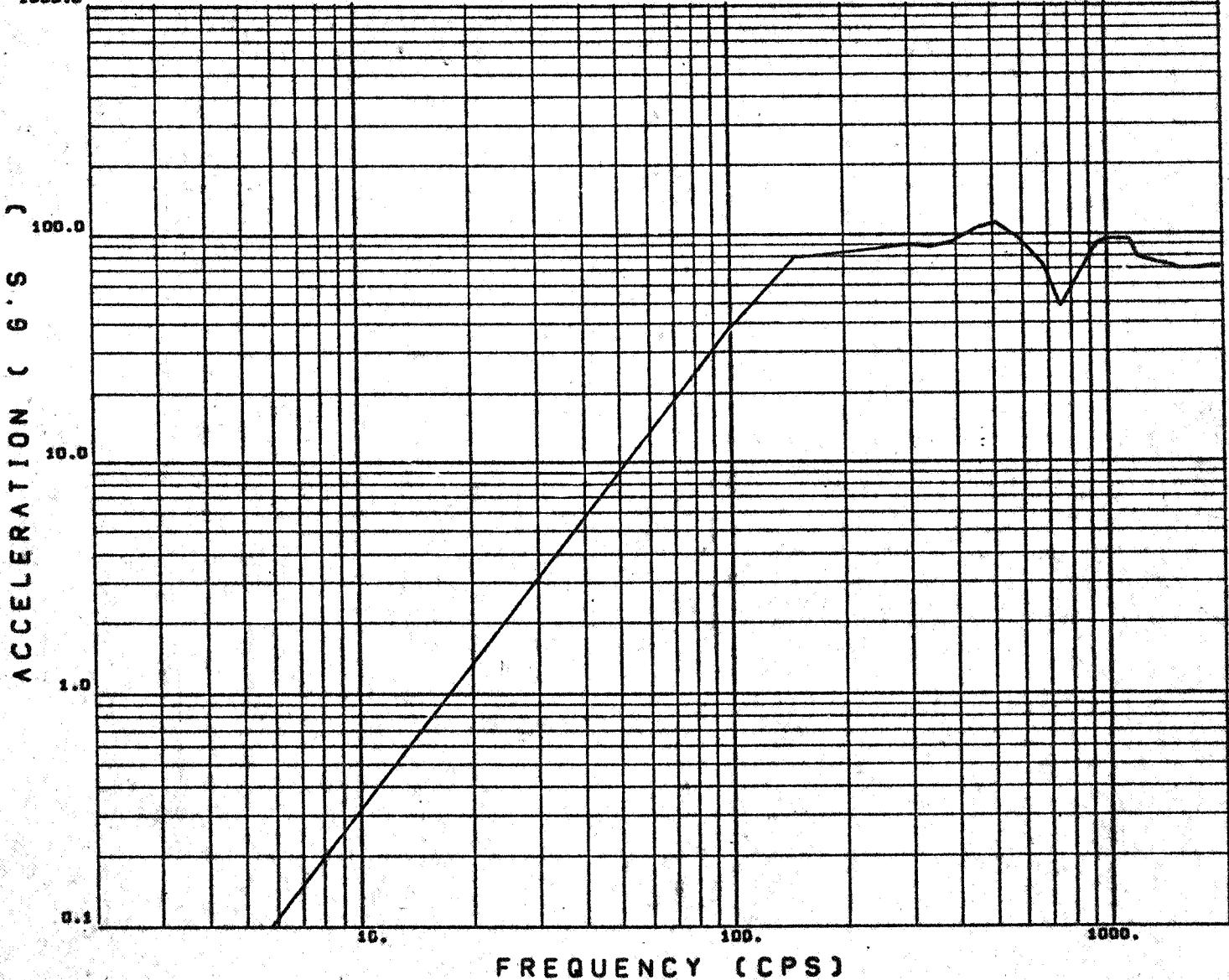
AXIS OF EXCITATION.... B

PICK UP NUMBER (1) ... 1 HB63

PICK UP RESPONSE..... B

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DOUGLAS AIRCRAFT COMPANY, INC.

PAGE NO. A9
REPORT NO. R 6033-2

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION ---

NOTE... SEE PAGE B3
FOR PICK UP LOCATION

COMMENT---

LEGEND...

UPSWEET -----

DOWNSWEEP -----

TEST CONDITIONS....

TEST DATE..... 6/16/67

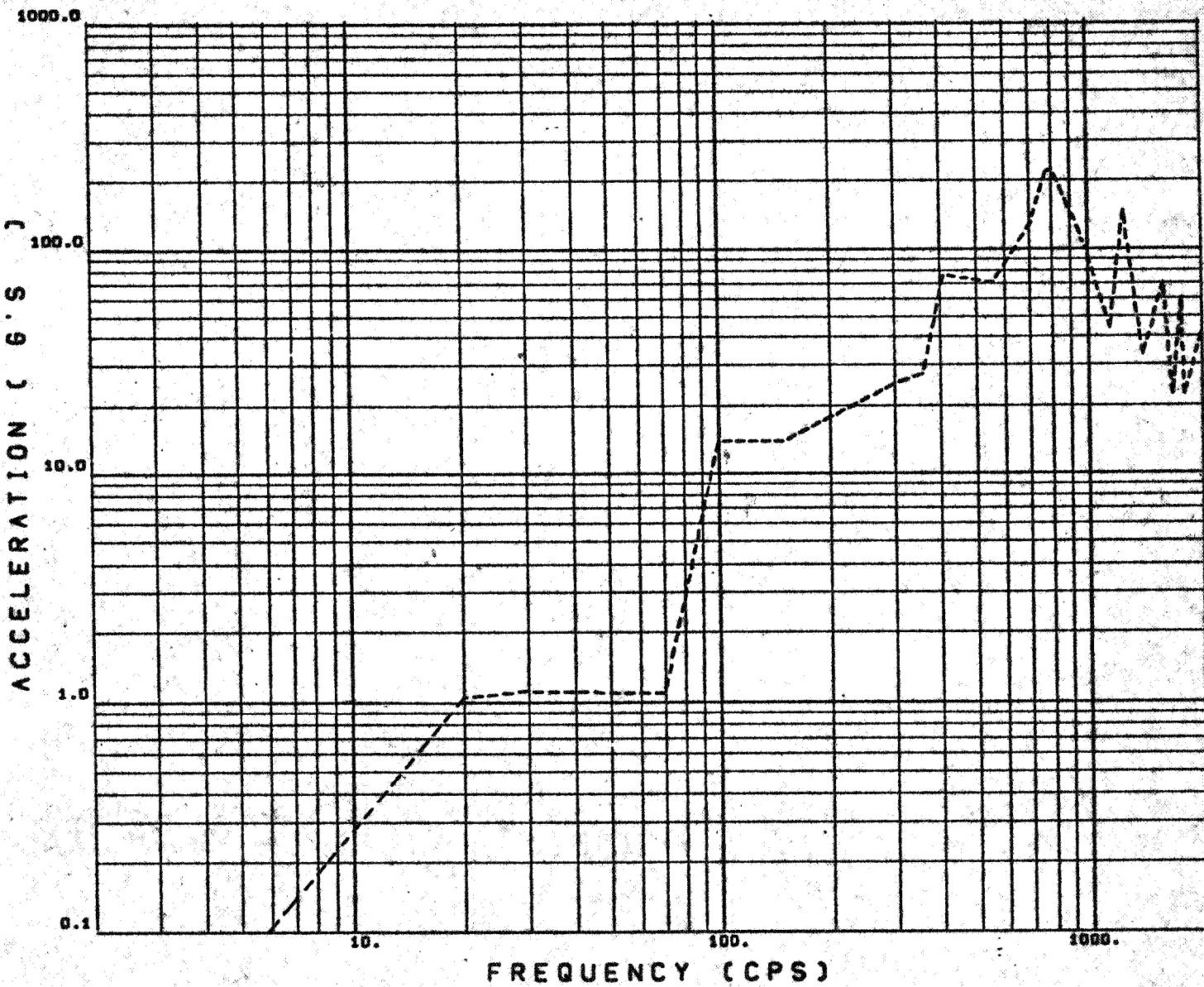
AXIS OF EXCITATION.... B

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PICK UP RESPONSE..... A

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A8



DOUGLAS AIRCRAFT COMPANY, INC.

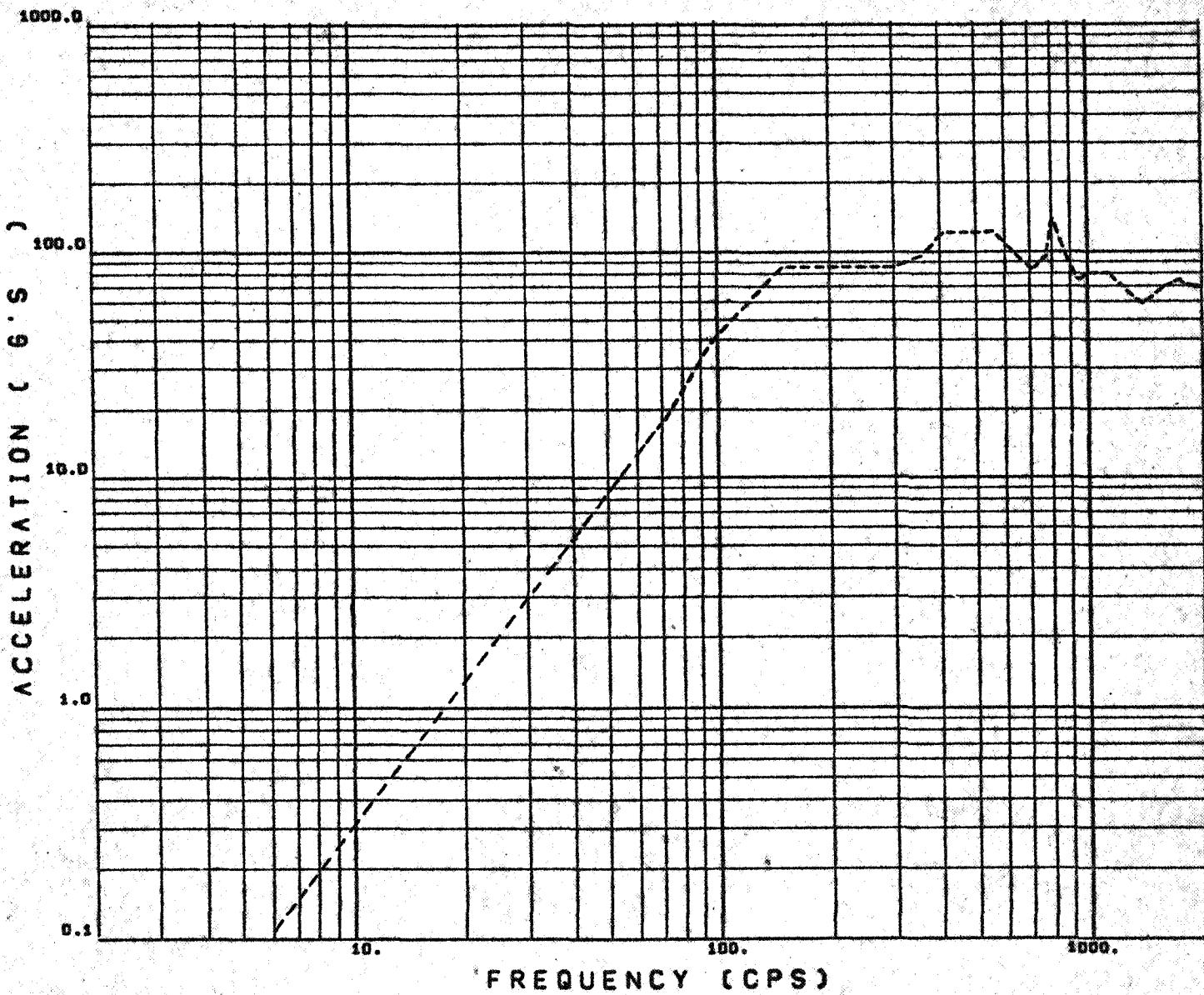
A10
PAGE NO.
REPORT NO. 6033-1

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION --- 
NOTE... SEE PAGE 
FOR PICK UP LOCATION
COMMENT---
LEGEND...
UPSWEEP -----
DOWNSWEEP -----

TEST CONDITIONS...

TEST DATE..... 6/16/67
AXIS OF EXCITATION.... B
PICK UP NUMBER (3) ... 3 NC59
PICK UP RESPONSE..... B
INPUT ACCEL.PER PAGE.. A8



DOUGLAS AIRCRAFT COMPANY, INC.

PAGE NO. A 11
REPORT NO. 86033-1

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION ---

NOTE... SEE PAGE 33
FOR PICK UP LOCATION

COMMENT---

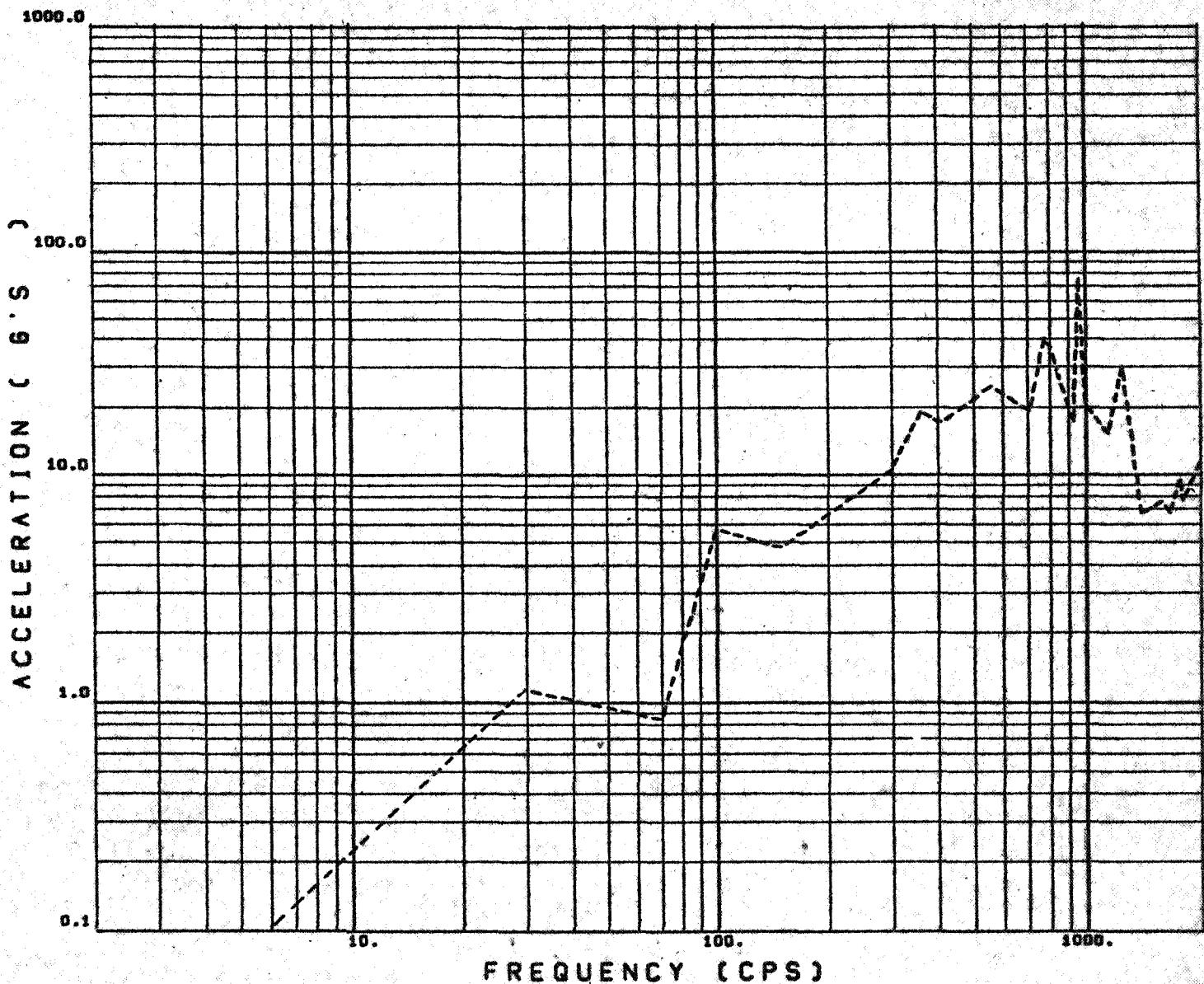
LEGEND...

UPSWEEP -----

DOWNSWEEP -----

TEST CONDITIONS...

TEST DATE..... 6/16/67
AXIS OF EXCITATION... B
PICK UP NUMBER (4)... 4 PB83
PICK UP RESPONSE..... C
INPUT ACCEL.PER PAGE... A8



DOUGLAS AIRCRAFT COMPANY, INC.

PAGE NO. A12
REPORT NO. 6033-1

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION ---

NOTE... SEE PAGE B5

FOR PICK UP LOCATION

COMMENT--- REFERENCE CHANNEL

LEGEND...

UPSHEEP -----

DOWNSHEEP -----

TEST CONDITIONS....

TEST DATE..... 6/17/67

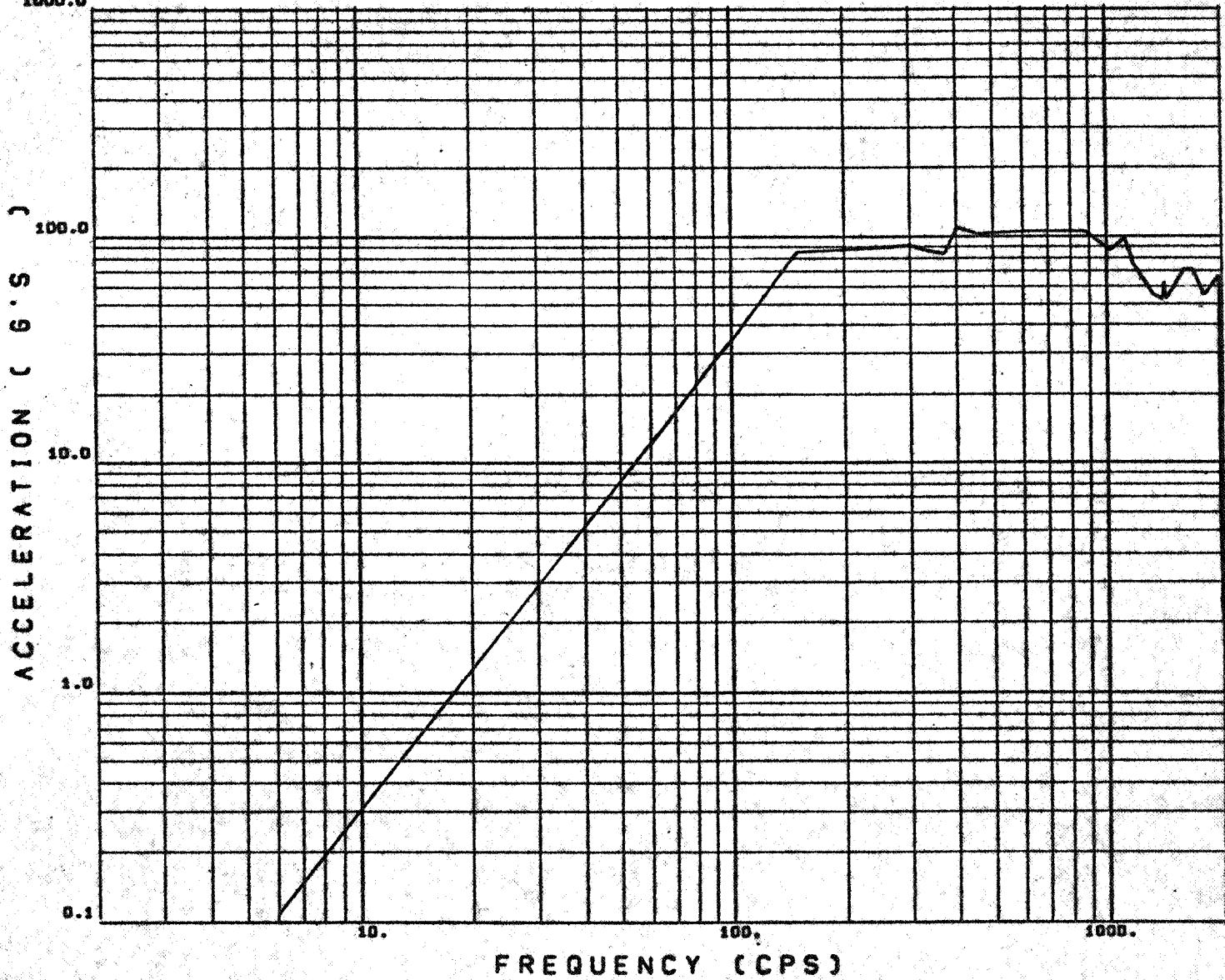
AXIS OF EXCITATION.... C

PICK UP NUMBER (1)... 1 HB63

PICK UP RESPONSE..... C

INPUT ACCEL.PER PAGE.. -----

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DOUGLAS AIRCRAFT COMPANY, INC.

A 13
PAGE NO.
REPORT NO. R6033-1

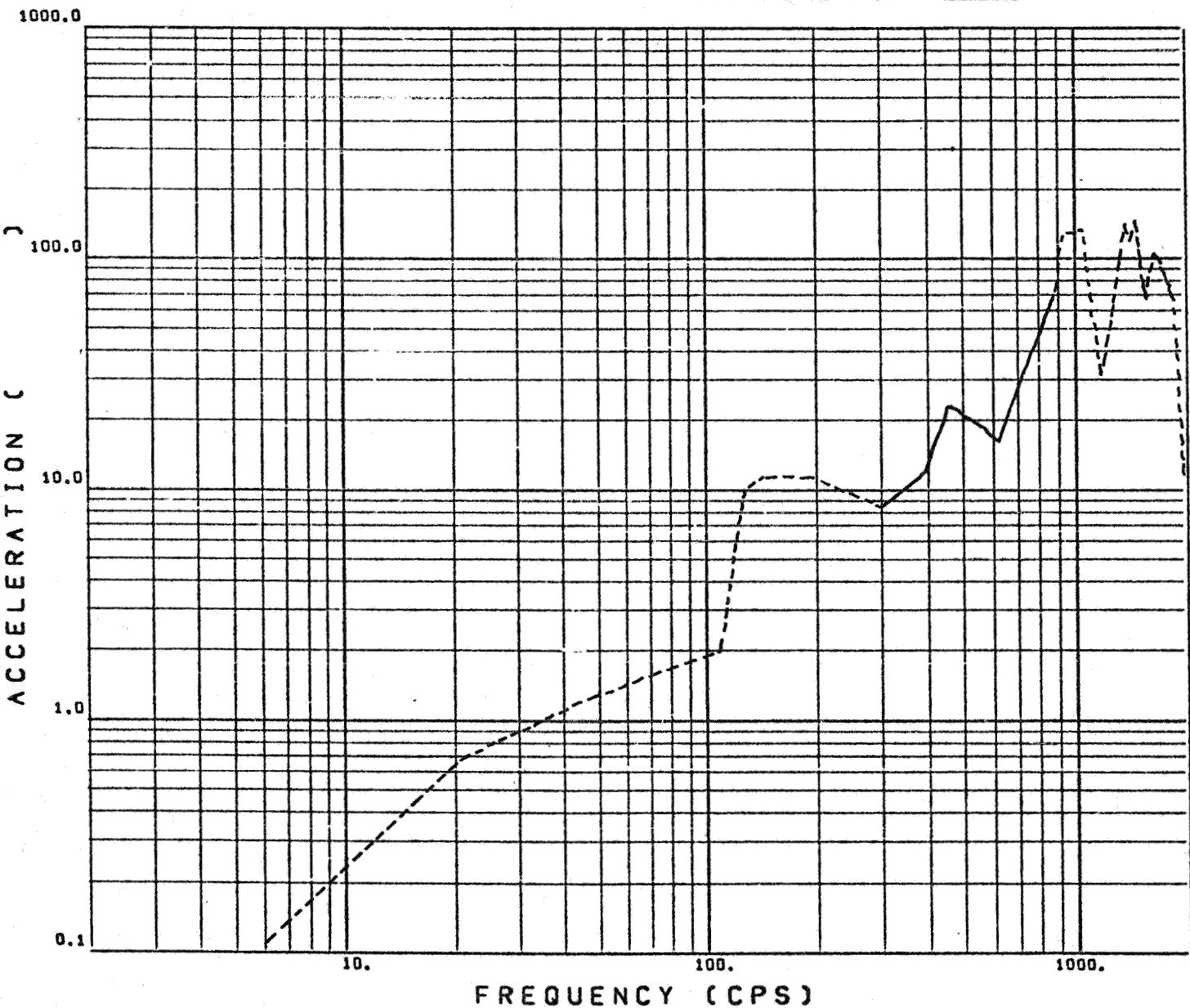
SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION --- B5
NOTE... SEE PAGE B5
FOR PICK UP LOCATION

LEGEND...
UPSWEEP ———
DOWNSWEEP -----

TEST CONDITIONS...

TEST DATE..... 6/17/67
AXIS OF EXCITATION.... C
PICK UP NUMBER (10).... 2 NC63
PICK UP RESPONSE..... A
INPUT ACCEL.PER PAGE.. A 12



DOUGLAS AIRCRAFT COMPANY, INC.

A 14

PAGE NO. R6038-1
REPORT NO.

SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION --- B5

NOTE... SEE PAGE FOR PICK UP LOCATION

COMMENT---

LEGEND...

UPSHEEP -----

DOWNSHEEP -----

TEST CONDITIONS....

TEST DATE..... 6/17/67

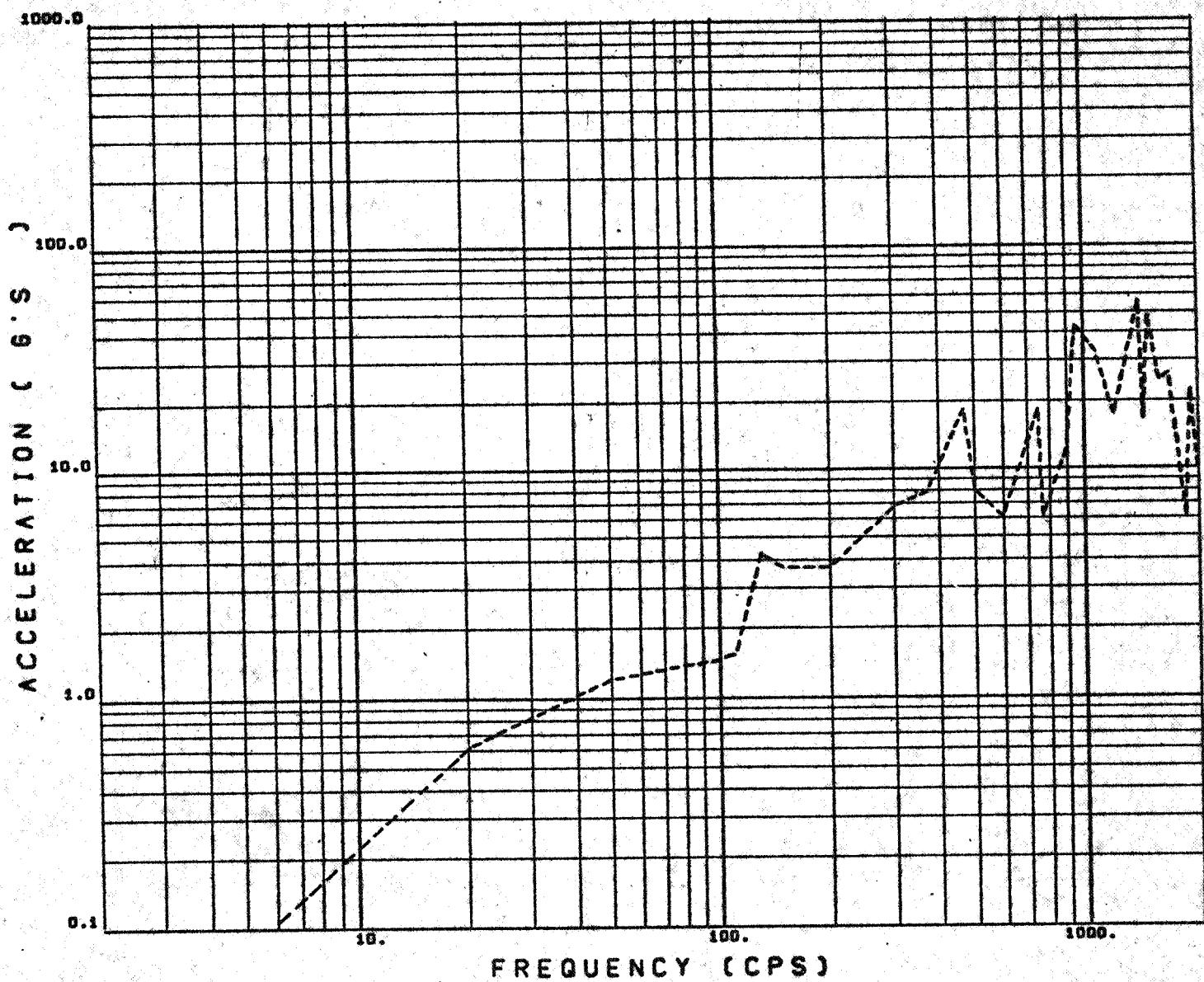
AXIS OF EXCITATION.... C

PICK UP NUMBER (3) ... 3 NC59

PICK UP RESPONSE..... B

INPUT ACCEL.PER PAGE..

A 12



DOUGLAS AIRCRAFT COMPANY, INC.

PAGE NO. A15
REPORT NO. 16030-1

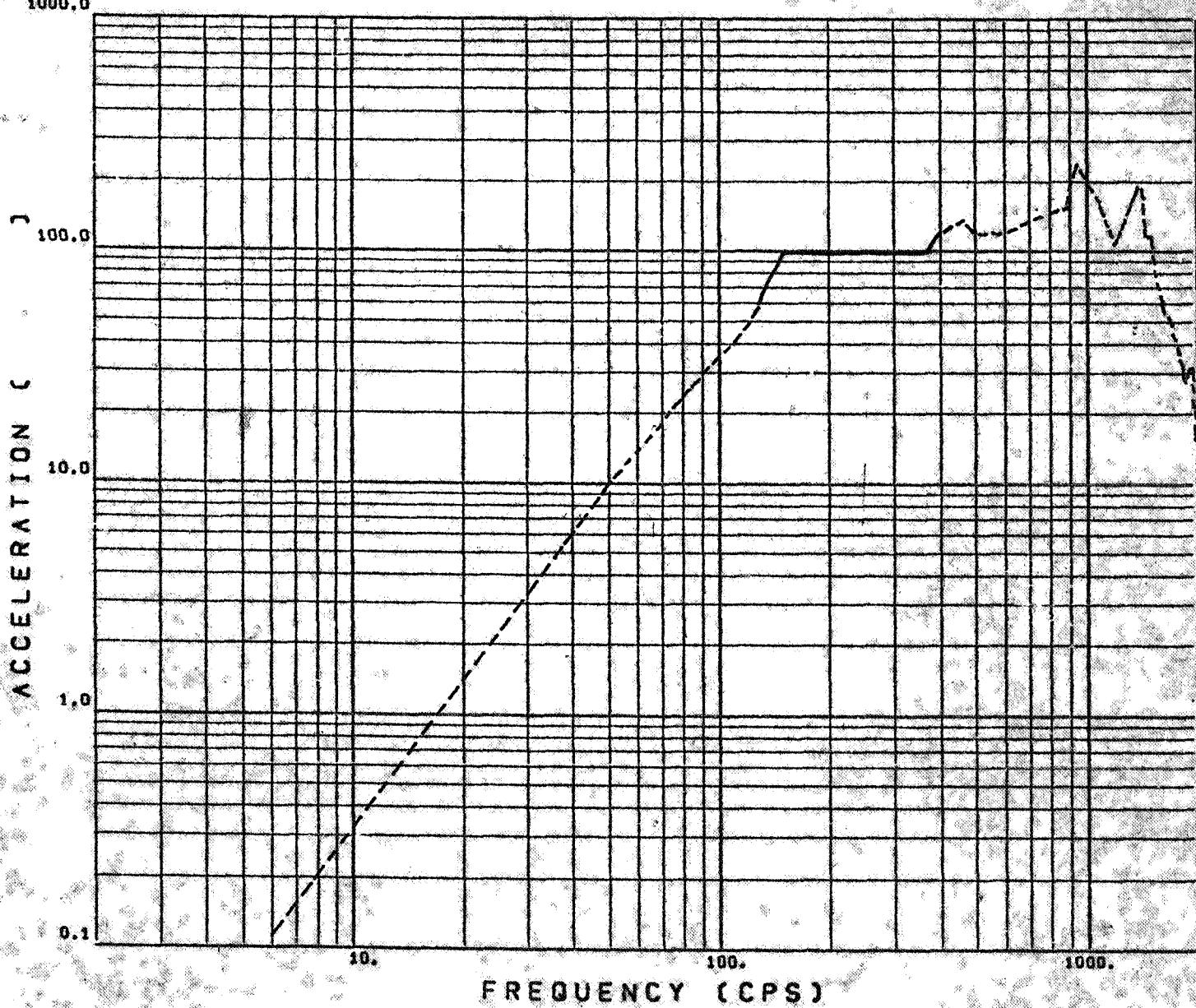
SINUSOIDAL FREQUENCY SWEEP
SIV-B VENT PORT CHECK VALVE
(J-20)

CONFIGURATION --- B5
NOTE... SEE PAGE B5
FOR PICK UP LOCATION

LEGEND...
UPSWEEP ———
DOWNSWEEP -----
1000.0

TEST CONDITIONS....

TEST DATE..... 6/17/67
AXIS OF EXCITATION.... C
PICK UP NUMBER (9) ... 4 ~~7003~~
PICK UP RESPONSE.... C
INPUT ACCEL.PER PAGE.. A12



DSV-1B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J - 20

CONFIGURATION

TEST CONDITIONS

P/N. 1B 57481-1

TEST DATE 6/17/1967

S/N. 9510411

AXIS OF EXCITATION 9

NOTE

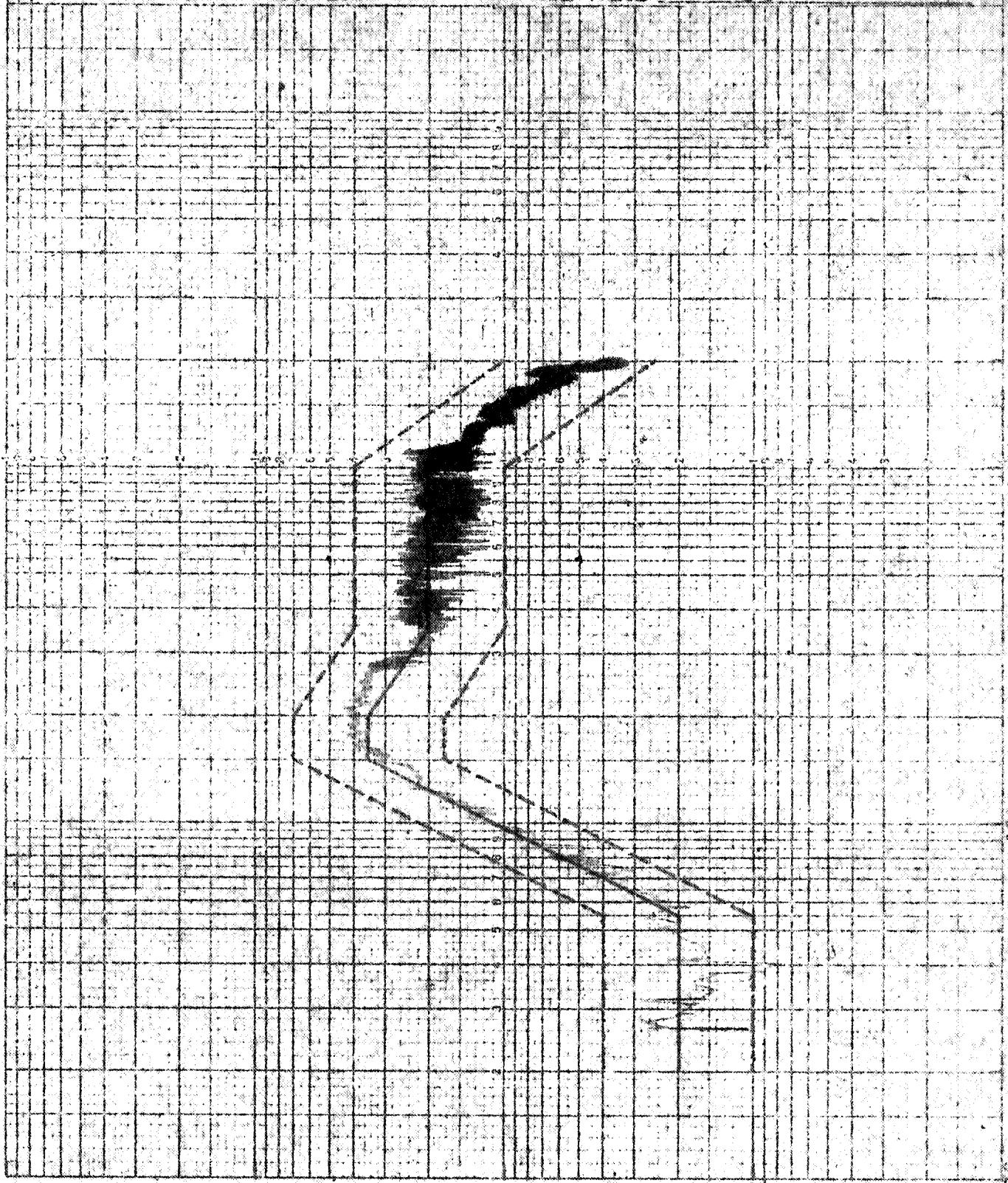
PICK-UP NUMBER 1

SEE PAGE B1 FOR
PICK-UP LOCATION

PICK-UP RESPONSE 1

INPUT ACCELERATION PER PAGE

RMS VALUE 5.5



DEV-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J - 20

CONFIGURATION

P/N 1B 67481-1

S/N 9510511

NOTE

SEE PAGE B3 FOR
PICK-UP LOCATION

TEST CONDITIONS

TEST DATE 6/17/1967

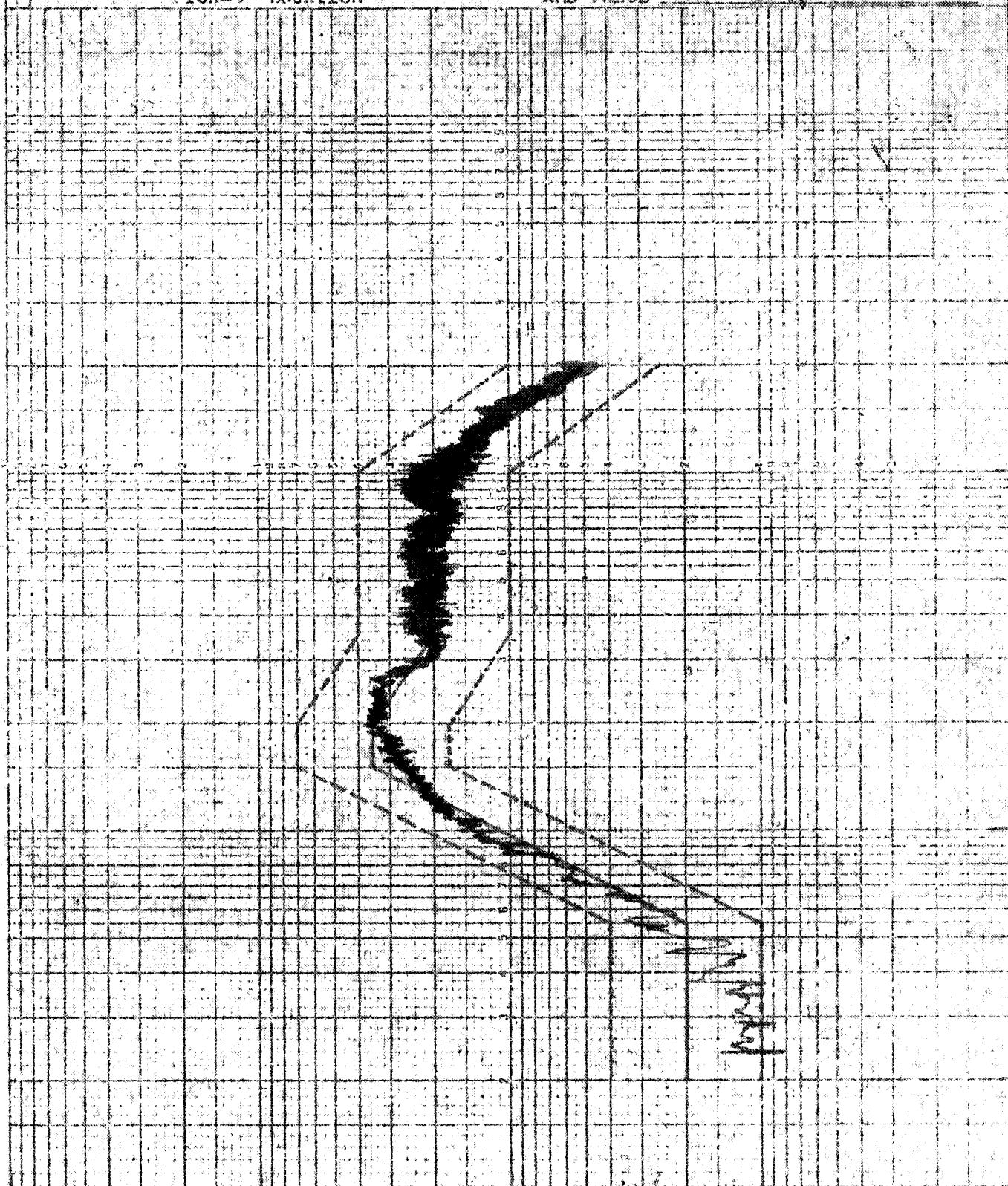
AXIS OF EXCITATION B

PICK-UP NUMBER 1

PICK-UP RESPONSE B

INPUT ACCELERATION PER PAGE

RMS VALUE 55



DSV-AB RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J - 20

CONFIGURATION

P/N 1B 67481-1

S/N 10 511

NOTE

SEE PAGE B5 FOR
PICK-UP LOCATION

TEST CONDITIONS

TEST DATE 6/17/1967

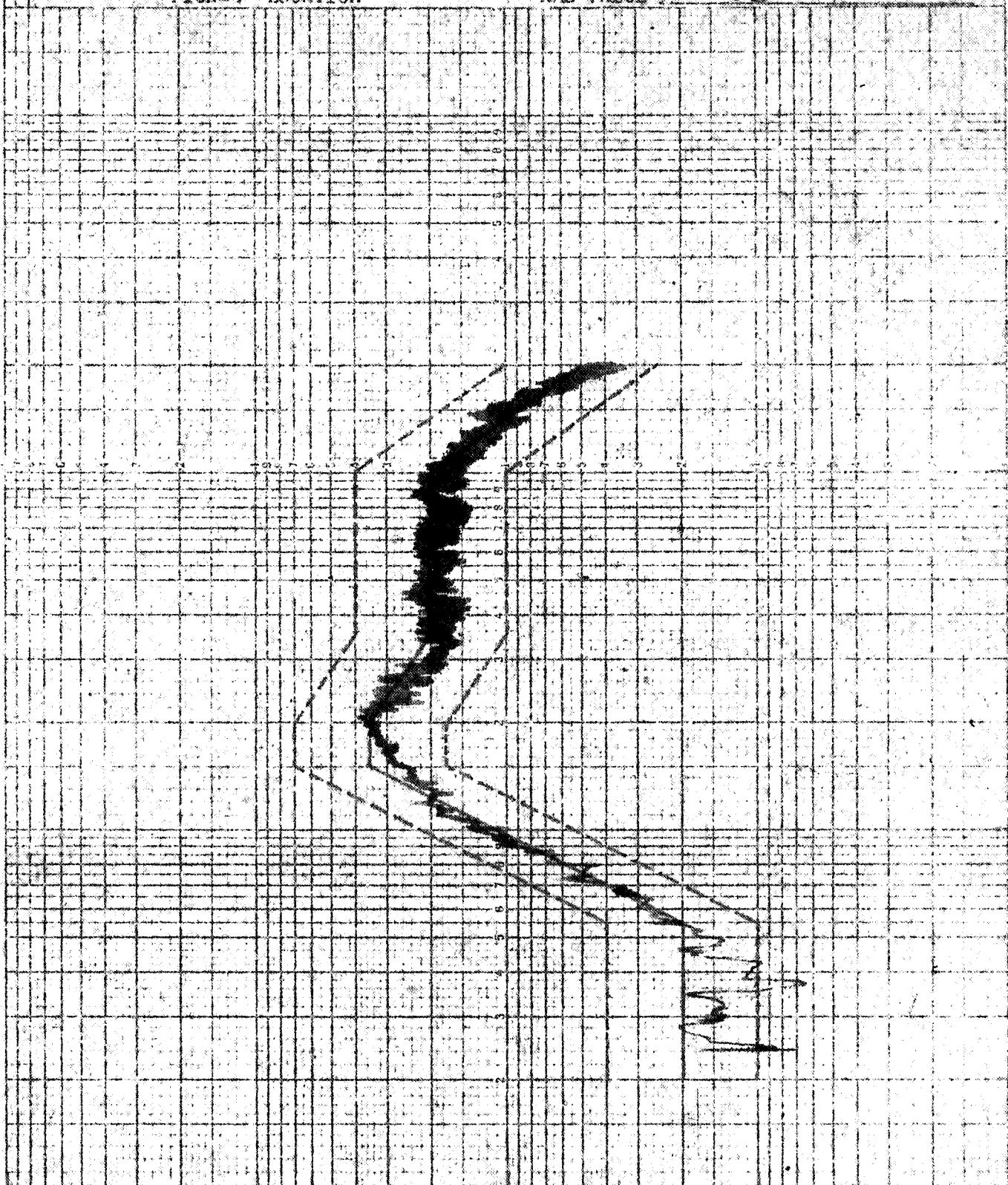
AXIS OF EXCITATION G

PICK-UP NUMBER 1

PICK-UP RESPONSE G

INPUT ACCELERATION PER PAGE

RMS VALUE 53



DSY-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE (J-20)

CONFIGURATION
P/N BEG. RUN

NOTE:

SEE PAGE D-1 FOR

PICK-UP LOCATION

TEST CONDITIONS

TEST DATE 6-17-67

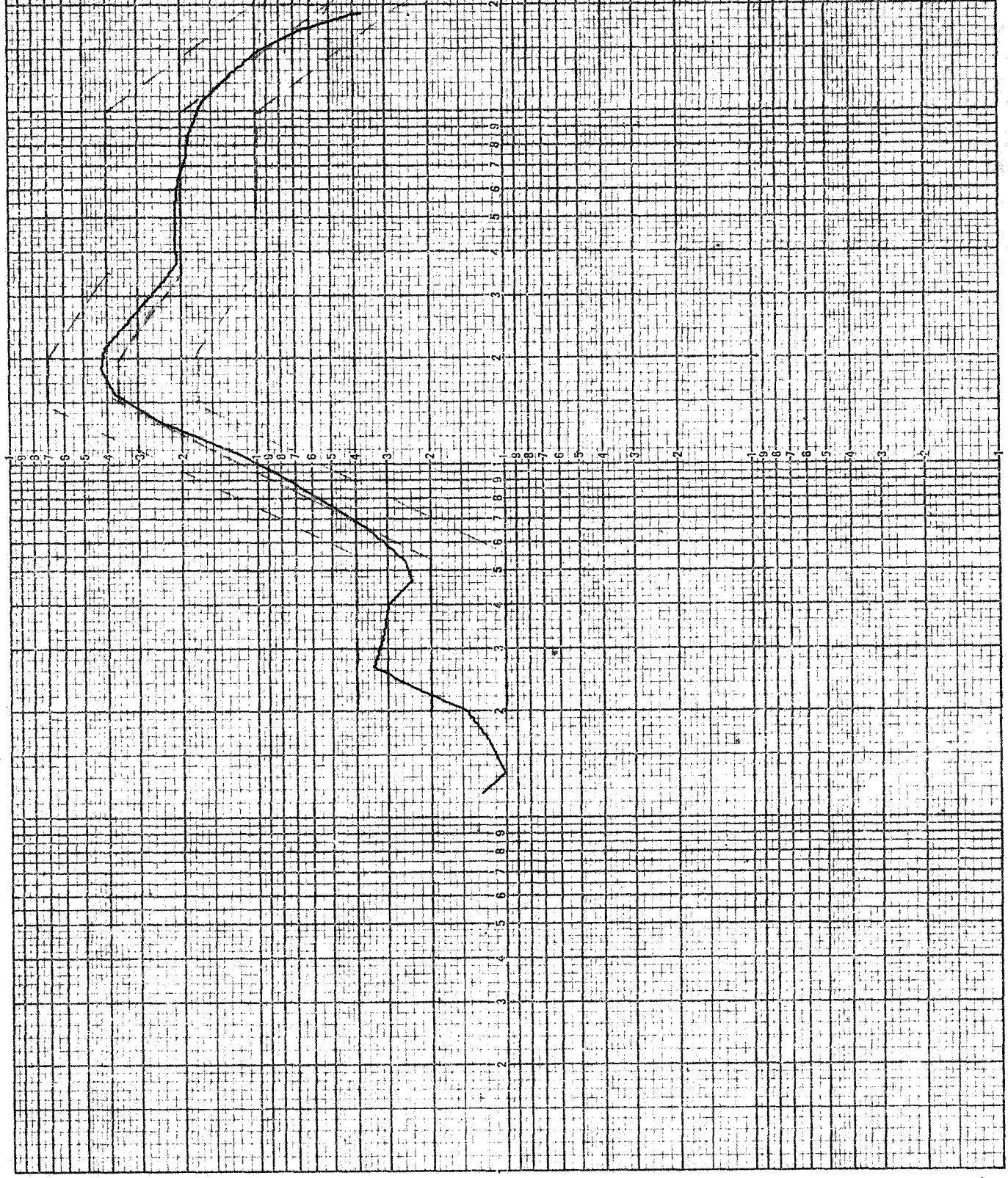
AXIS OF EXCITATION A

PICK-UP NUMBER 1

PICK-UP RESPONSE R

INPUT ACCELERATION PER PAGE

RMS VALUE 56.2



DSY-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE 1J-201

CONFIGURATION
P/N BEG. RUNNOTE
SEE PAGE D
PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE 6-17-67

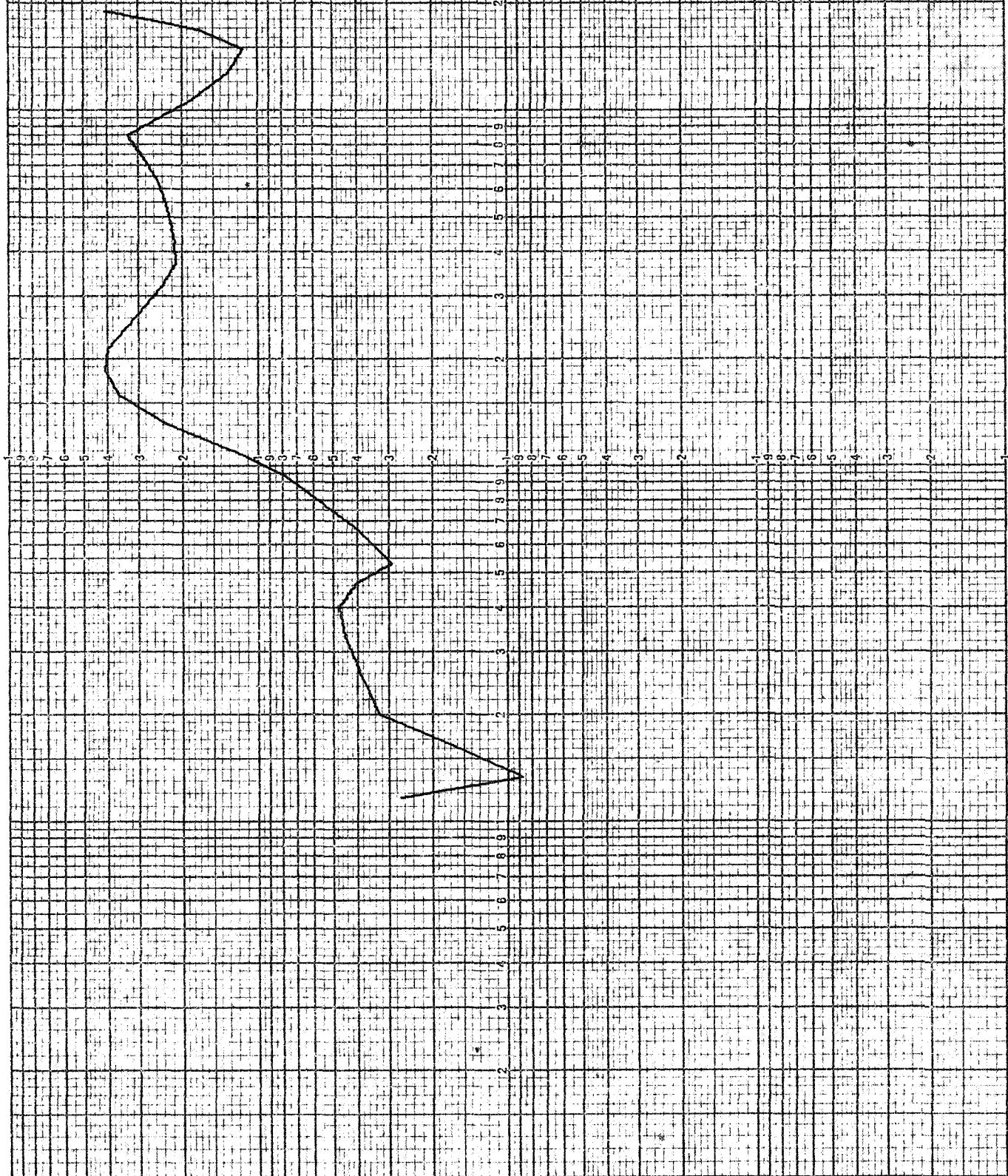
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PICK-UP NUMBER 2

PICK-UP RESPONSE R

INPUT ACCELERATION PER PAGE A 19

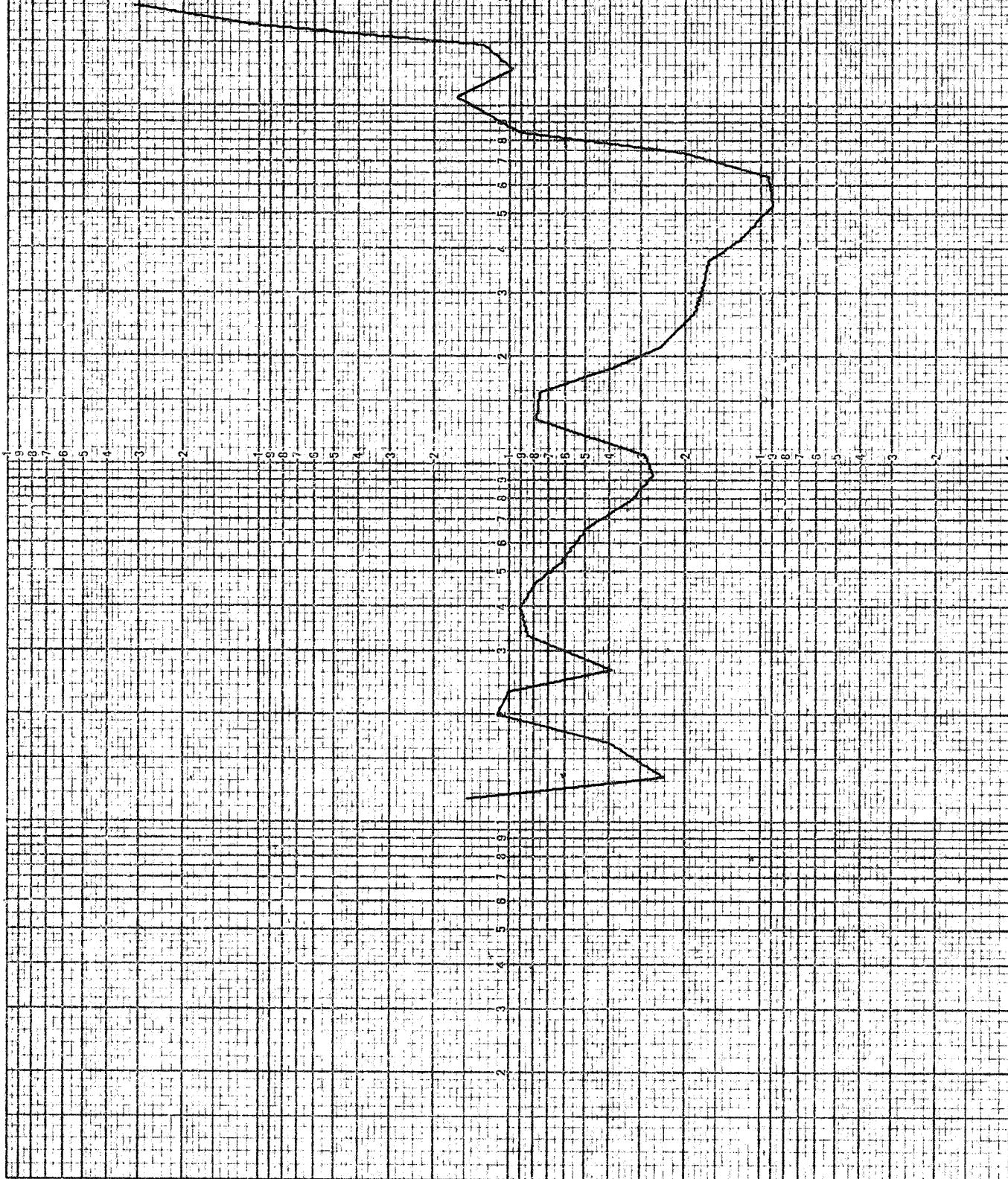
PMS VALUE 74.3



DSV-4B RANDOM VIBRATION TEST
VENT PORT CHECK VALVE (J-20)CONFIGURATION
P/N BEG. RUNNOTE
SEE PAGE D FOR
PICK-UP LOCATION

TEST CONDITIONS

TEST DATE 6-17-67
 AXIS OF EXCITATION X
 PICK-UP NUMBER 3
 PICK-UP RESPONSE B
 INPUT ACCELERATION PER PAGE A 13
 RMS VALUE 12.1



DSY-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE (J-20)

CONFIGURATION

P/N BEG. RUN

NOTE

SEE PAGE

B

PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE

6-17-67

Hertz of excitation

A

PICK-UP NUMBER

4

PICK-UP RESPONSE

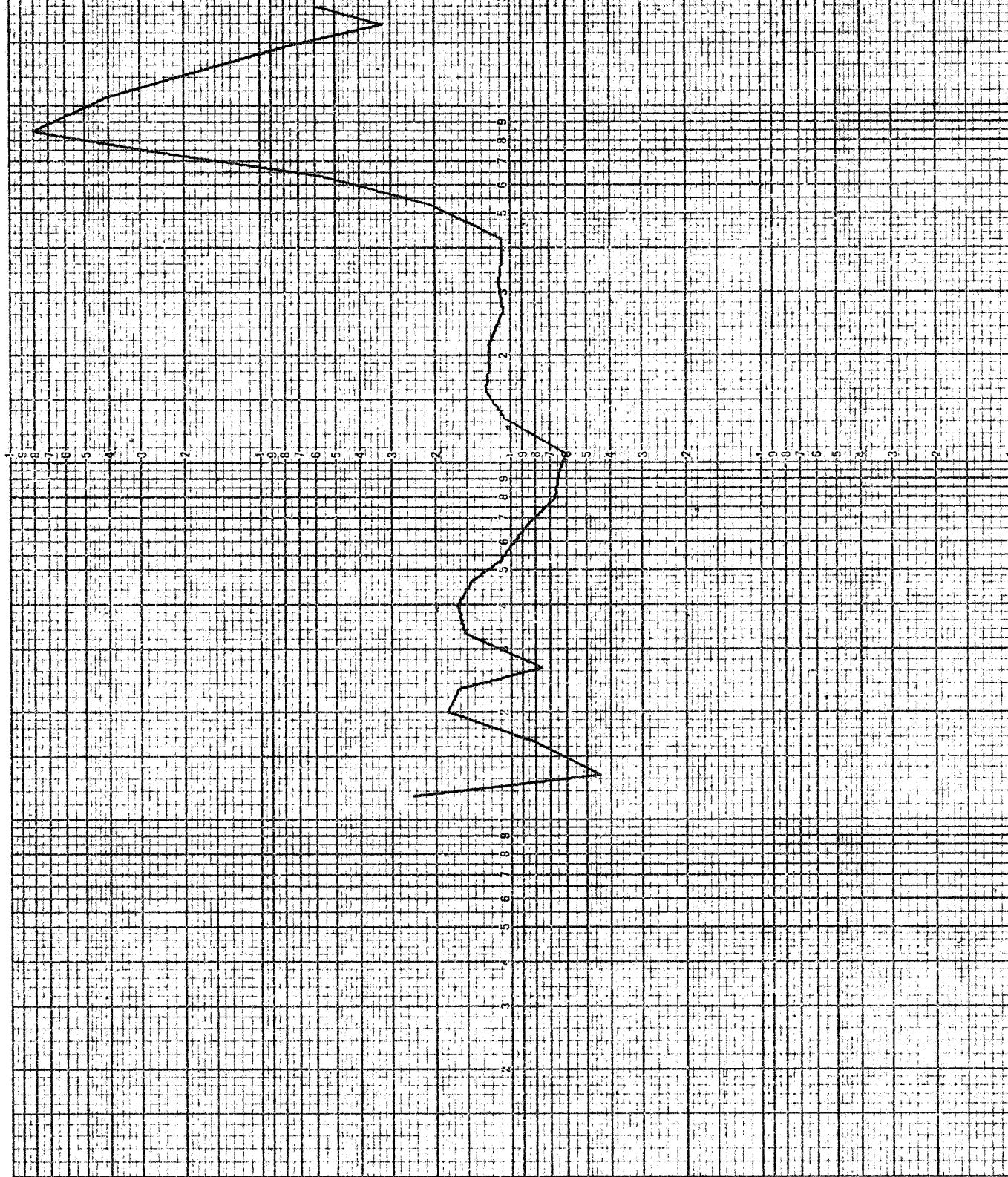
C

INPUT ACCELERATION PER PAGE

A 19

RMS VALUE

18.6



DSV-40 RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J-20

CONFIGURATION

P/N 1BETHB1-I
MIDDLE OF RONNOTE **D** SEE PAGE **3**
PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE

REV E-21-1

6-17-67

PXTS OF EXCITATION

B

PICK-UP NUMBER

1

PICK-UP RESPONSE

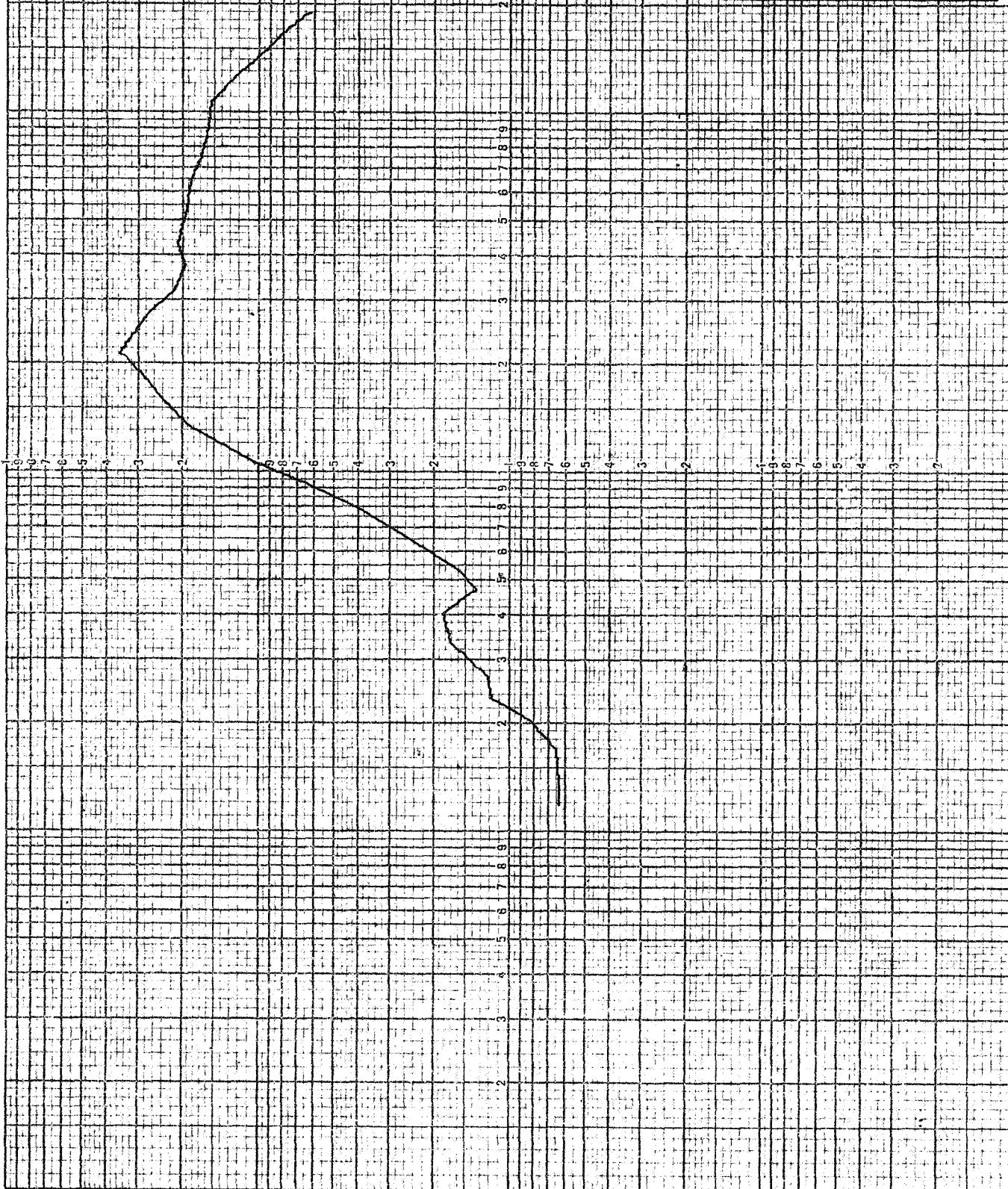
B

INPUT ACCELERATION PER PAGE

B

RMS VALUE

56.9



10.0

1.00

SPECTRAL DENSITY IN G²/HERTZ

.1000

.00100

1000.0

100.0
10.0
1.0

DSV-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J-2B

CONFIGURATION

P/N 18674B1-1

NOTE

SEE PAGE

PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE

6-17-67

AXIS OF EXCITATION

B

PICK-UP NUMBER

2

PICK-UP RESPONSE

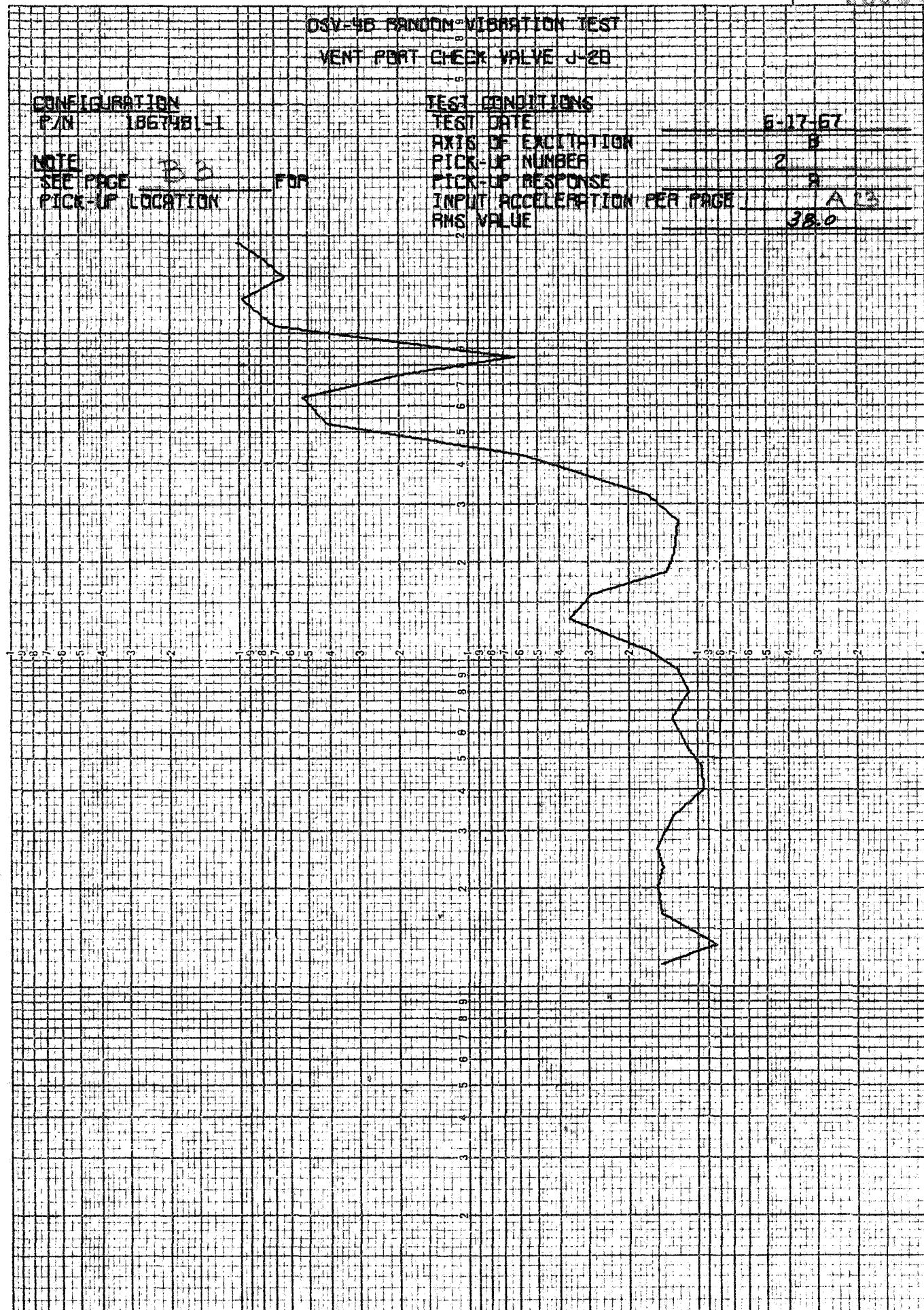
A

INPUT ACCELERATION PER PAGE

A

RMS VALUE

38.0



OSV-4D RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J-20

CONFIGURATION

P/N 1B67481-1

NOTE

SEE PAGE

PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE

6-17-67

AXIS OF EXCITATION

B

PICK-UP NUMBER

3

PICK-UP RESPONSE

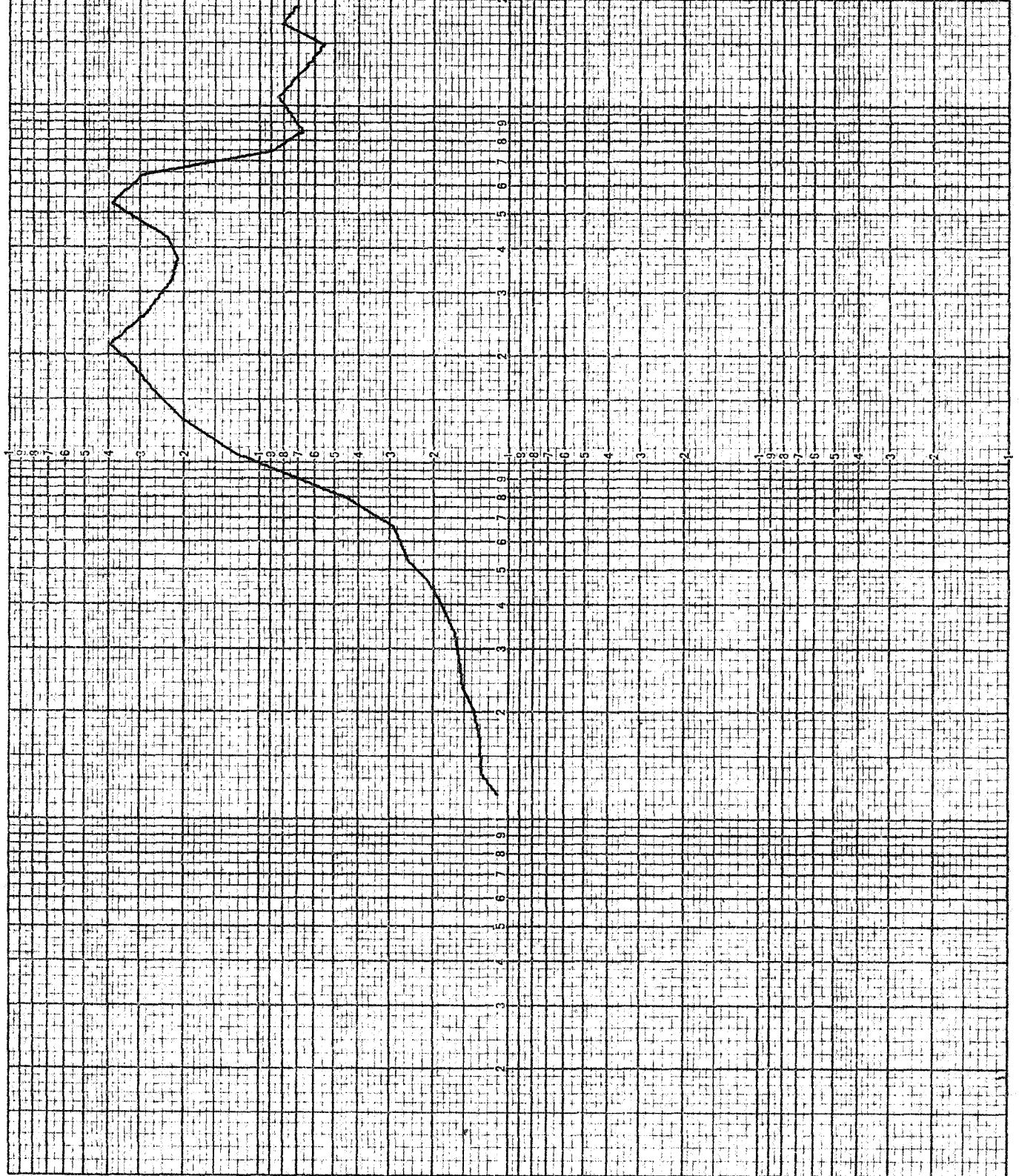
B

INPUT ACCELERATION PER PAGE

A 23

RMS VALUE

53.5



1000.0

FREQUENCY CPM

10.0

10.0

DSV-4B RANDOM VIBRATION TEST

VENT PORT CHEER VALVE J-20

CONFIGURATION
F/A N 1857481-1NOTE
SEE PAGE FOR
PICK-UP LOCATION

TEST CONDITIONS

TEST DATE 5-17-67

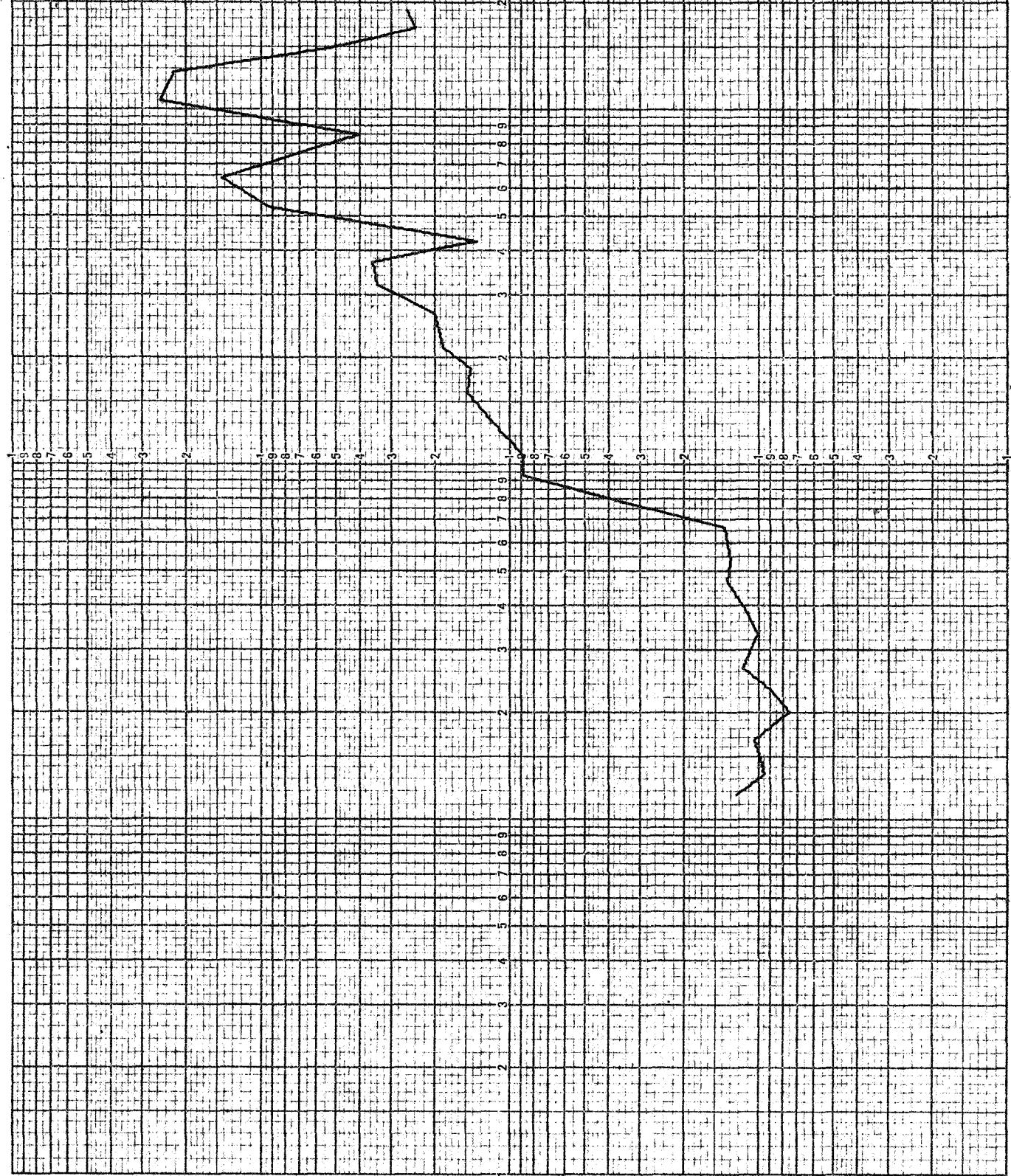
AXIS OF EXCITATION B

PICK-UP NUMBER 4

PICK-UP RESPONSE C

INPUT ACCELERATION PER PAGE A 2%

RMS VALUE 14.1



DSY-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J20

CONFIGURATION

P/N 186748L-1

NOTE

SEE PAGE 12

PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE

06-17-67

AXIS OF EXCITATION

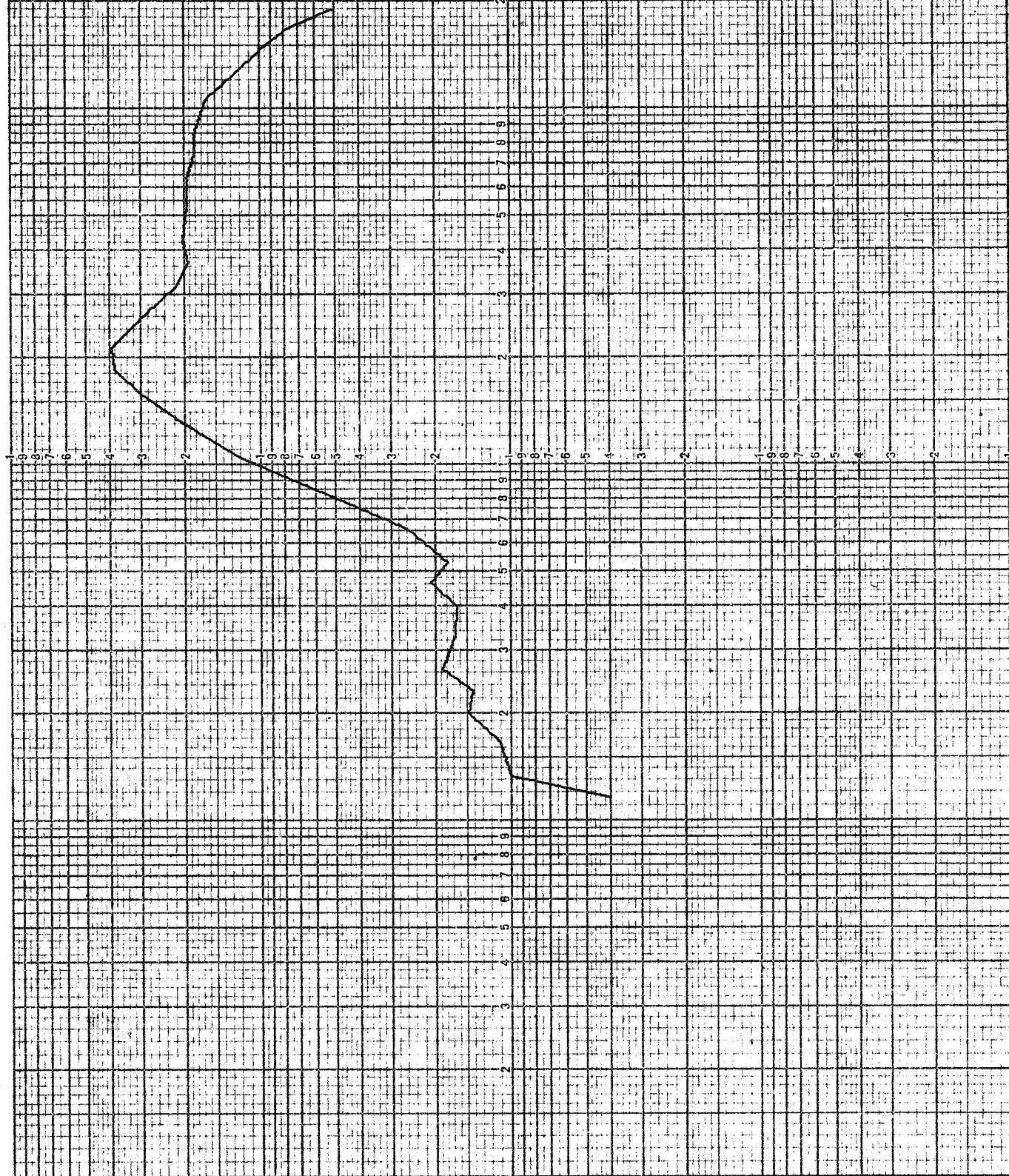
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PICK-UP RESPONSE

INPUT ACCELERATION PER PAGE

RMS VALUE

57.4



DSY-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J20

CONFIGURATION

P/N 1867481-1

NOTE

SEE PAGE 25

PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE

05-17-67

AXIS OF EXCITATION

C

PICK-UP NUMBER

2

PICK-UP RESPONSE

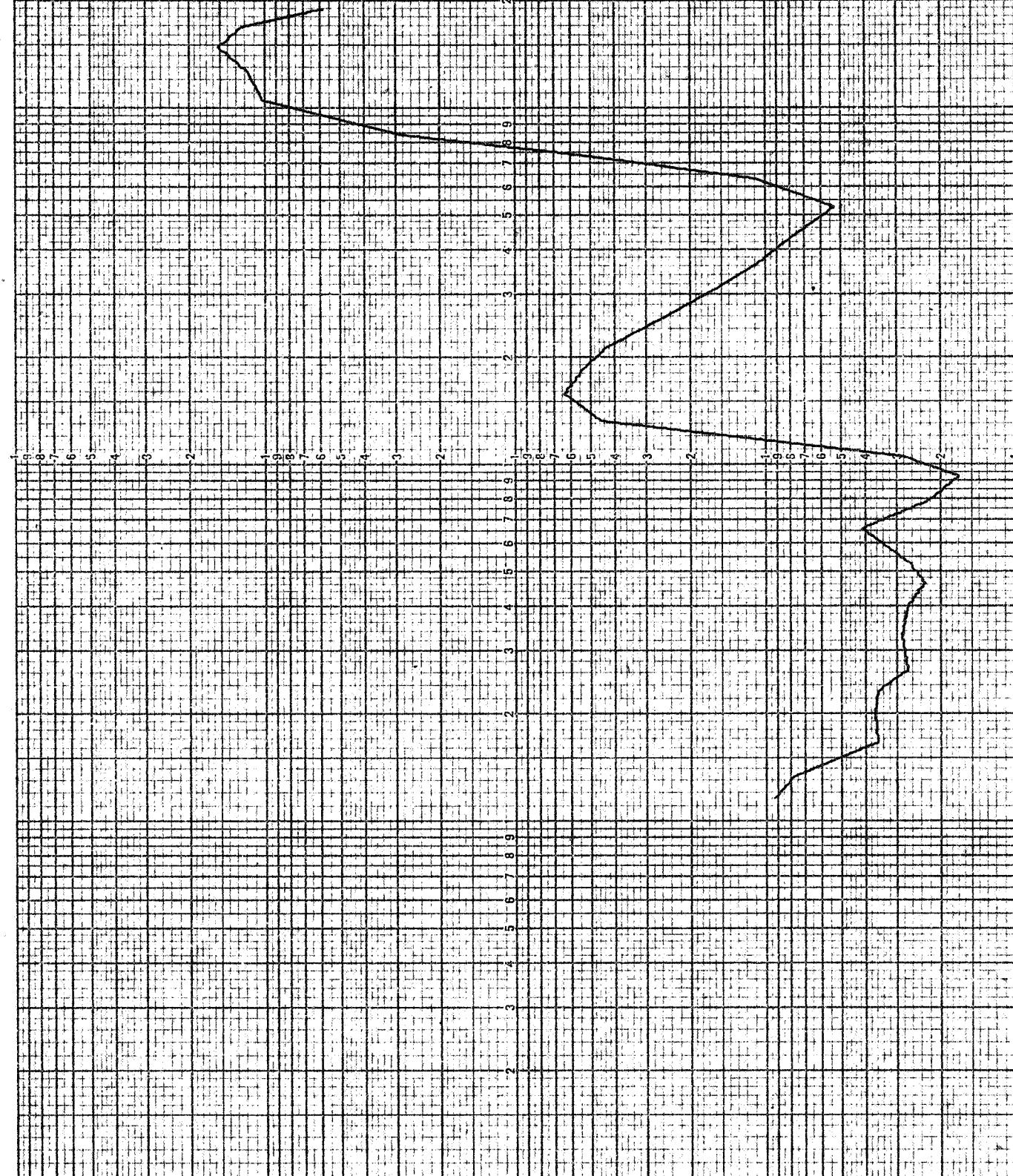
A

INPUT ACCELERATION PER PAGE

A-27

RMS VALUE

37.2



10.0

100.0

SPECTRAL DENSITY IN G²/HERTZ

.0100

.100

1.00

.00100

1000.0

100.0

100.0
FREQUENCY CPS

1000.0

DSV-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J20

CONFIGURATION

P/N 1867461-1

NOTE

SEE PAGE 5 FOR
PICK-UP LOCATION

TEST CONDITIONS

TEST DATE DE-17-67

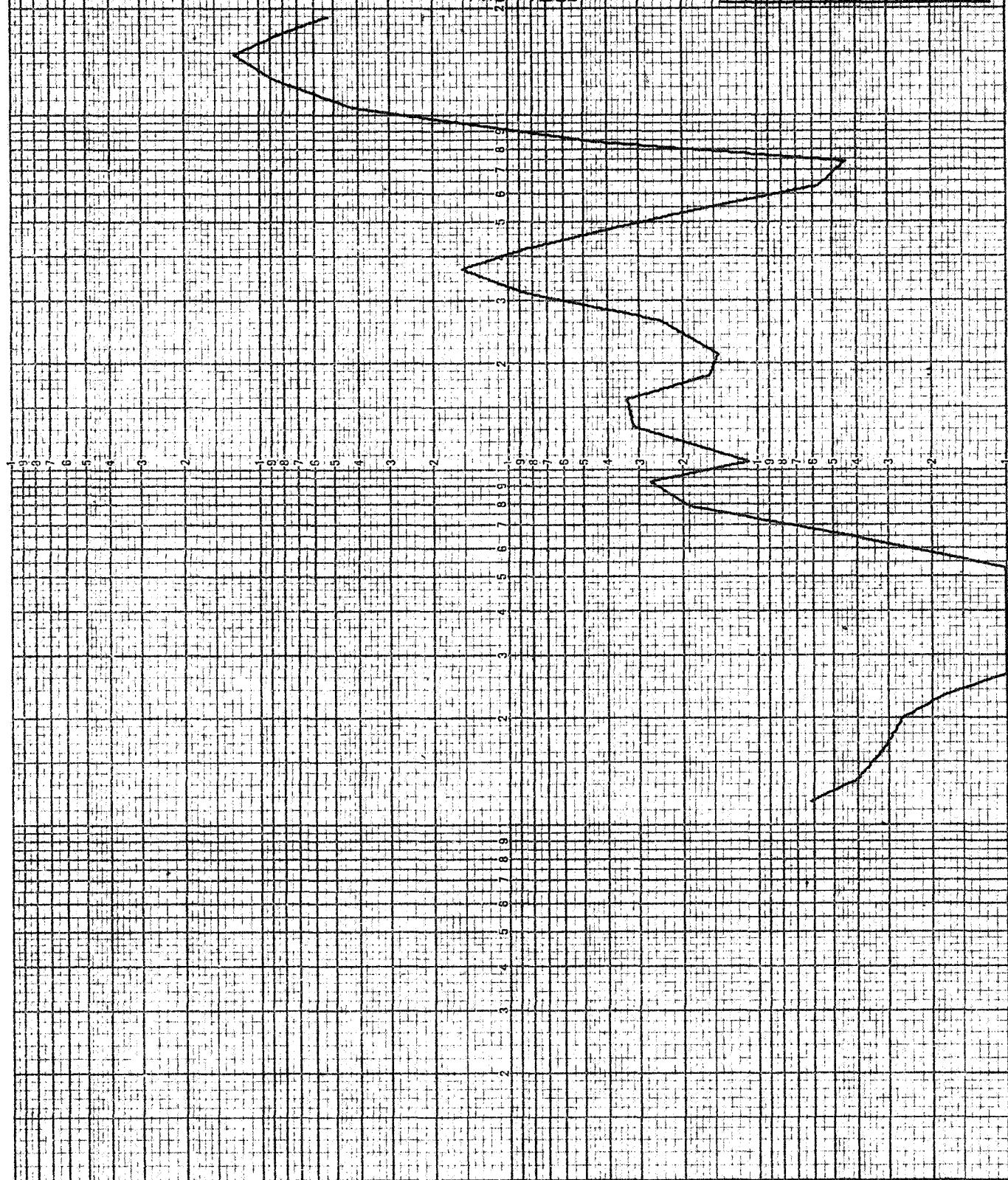
AXIS OF EXCITATION 3

PICK-UP NUMBER 3

PICK-UP RESPONSE 2

INPUT ACCELERATION PER PAGE A 27

RMS VALUE 1/4 G



USV-4B RANDOM VIBRATION TEST

VENT PORT CHECK VALVE J20

CONFIGURATION

P/N 18674BL-1

NOTE

SEE PAGE

PICK-UP LOCATION

FOR

TEST CONDITIONS

TEST DATE

DE-17-67

AXIS OF EXCITATION

PICK-UP NUMBER

4

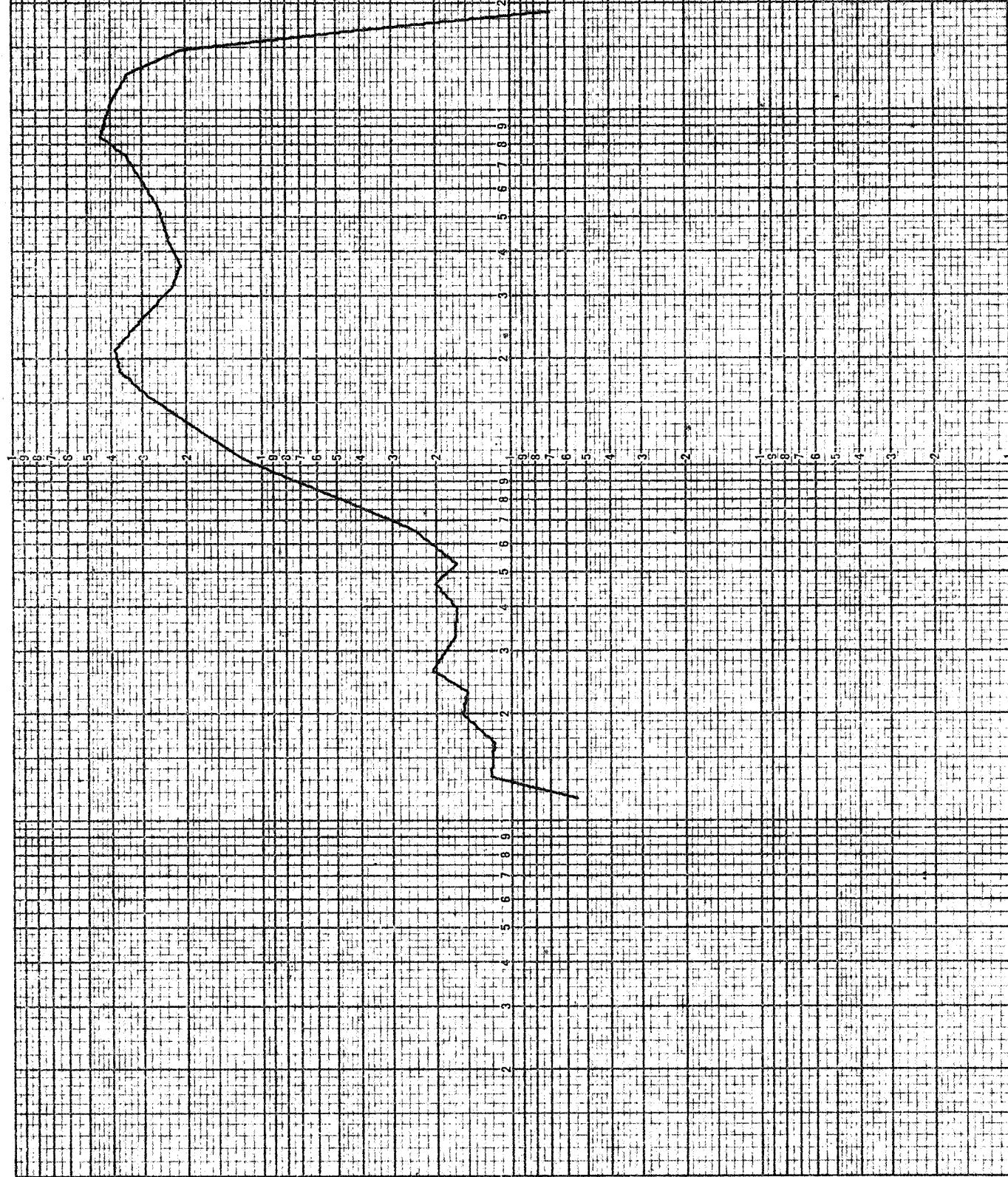
PICK-UP RESPONSE

INPUT ACCELERATION PER PAGE

27

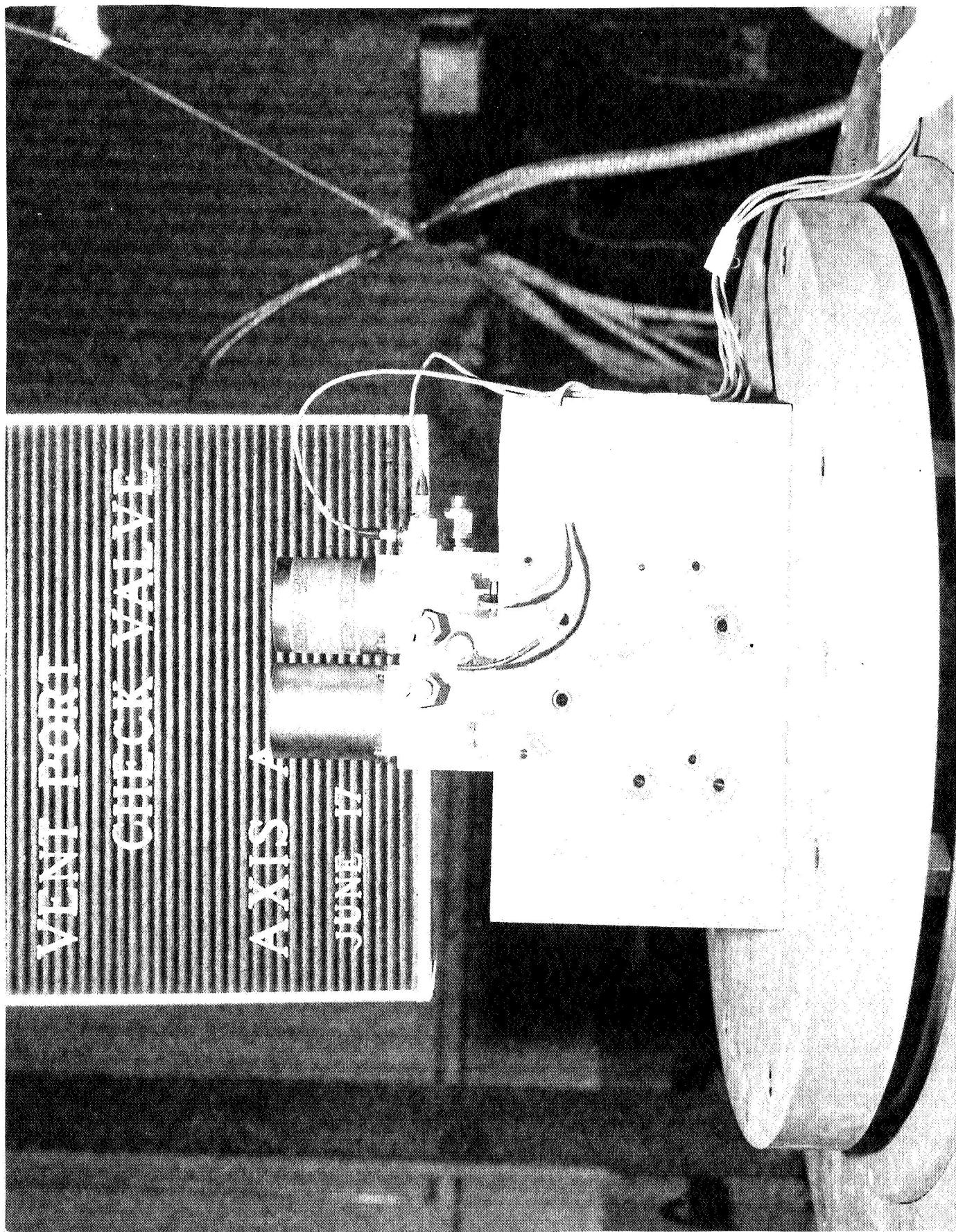
RMS VALUE

69.7



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MODEL DSV-4B

DOUGLAS AIRCRAFT COMPANY, INC. PAGE P 1
REPORT NO. R 5033-1

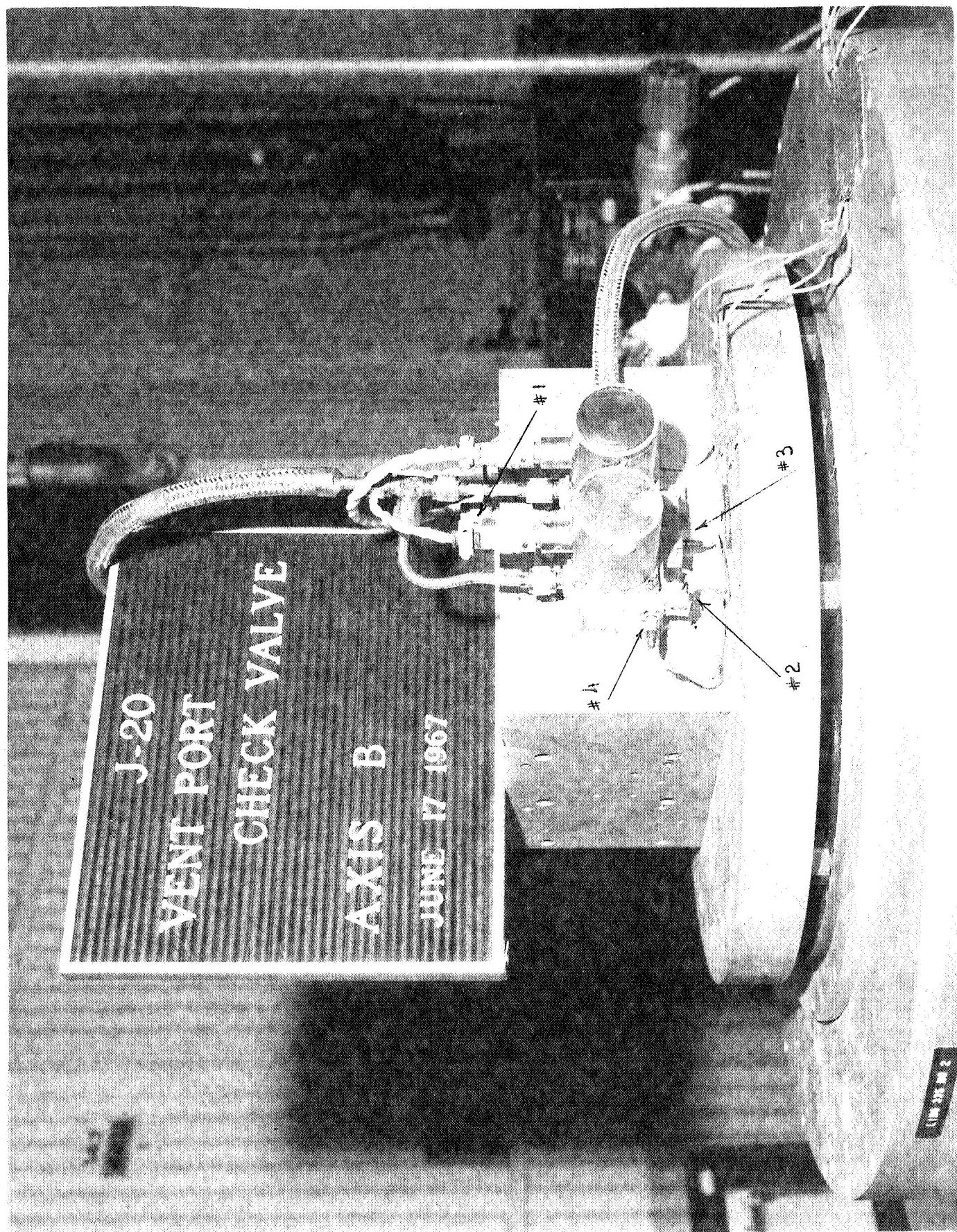


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DOUGLAS AIRCRAFT COMPANY, INC. PAGE B 2

MODEL DSV-4B

REPORT NO. R 6033-1



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N. Dev

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PAGE

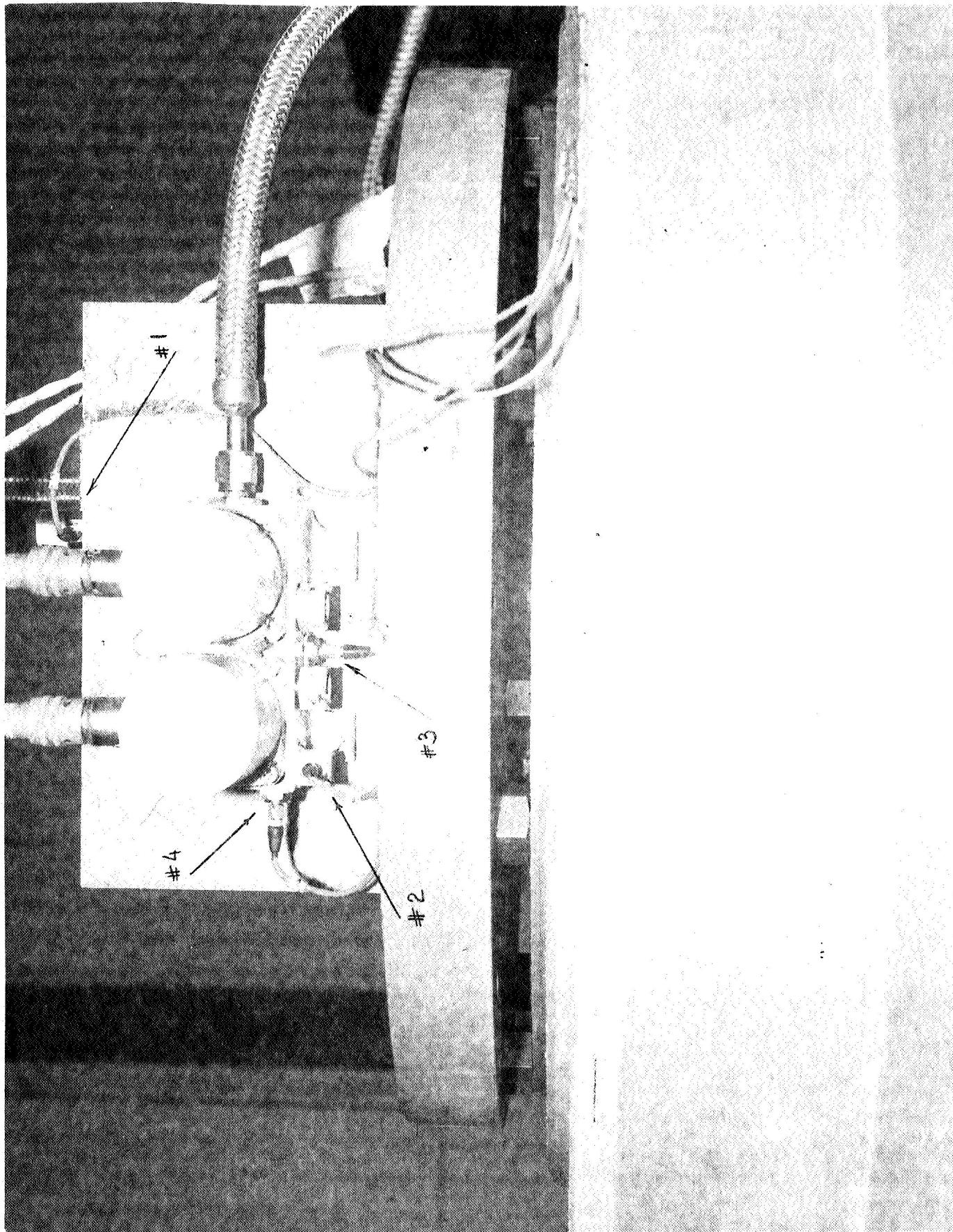
33

MODEL

DSV-4B

REPORT NO.

R 6024-3



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DOUGLAS AIRCRAFT COMPANY, INC.

PAGE

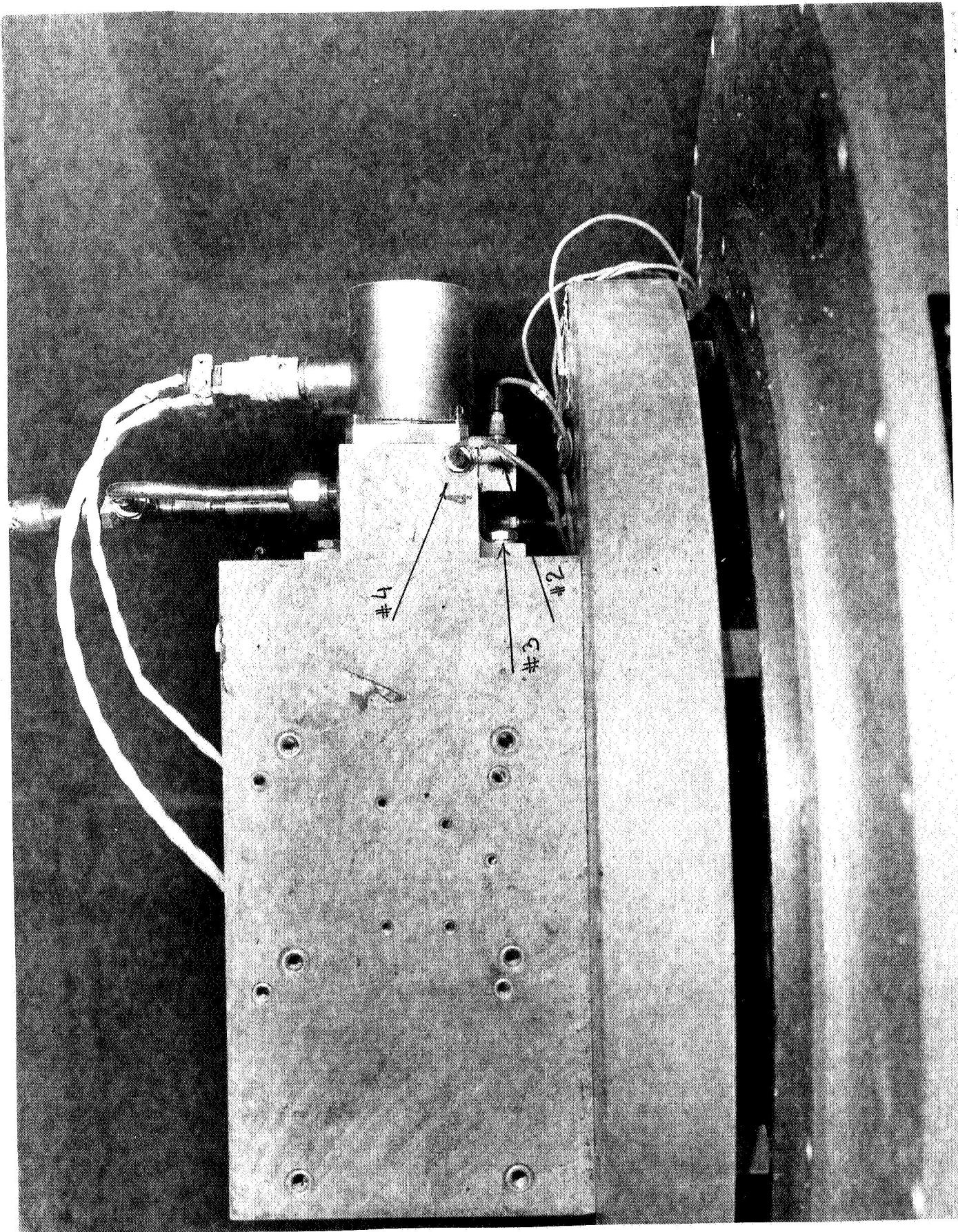
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MODEL

DSV-4E

REPORT NO.

R 6033-1



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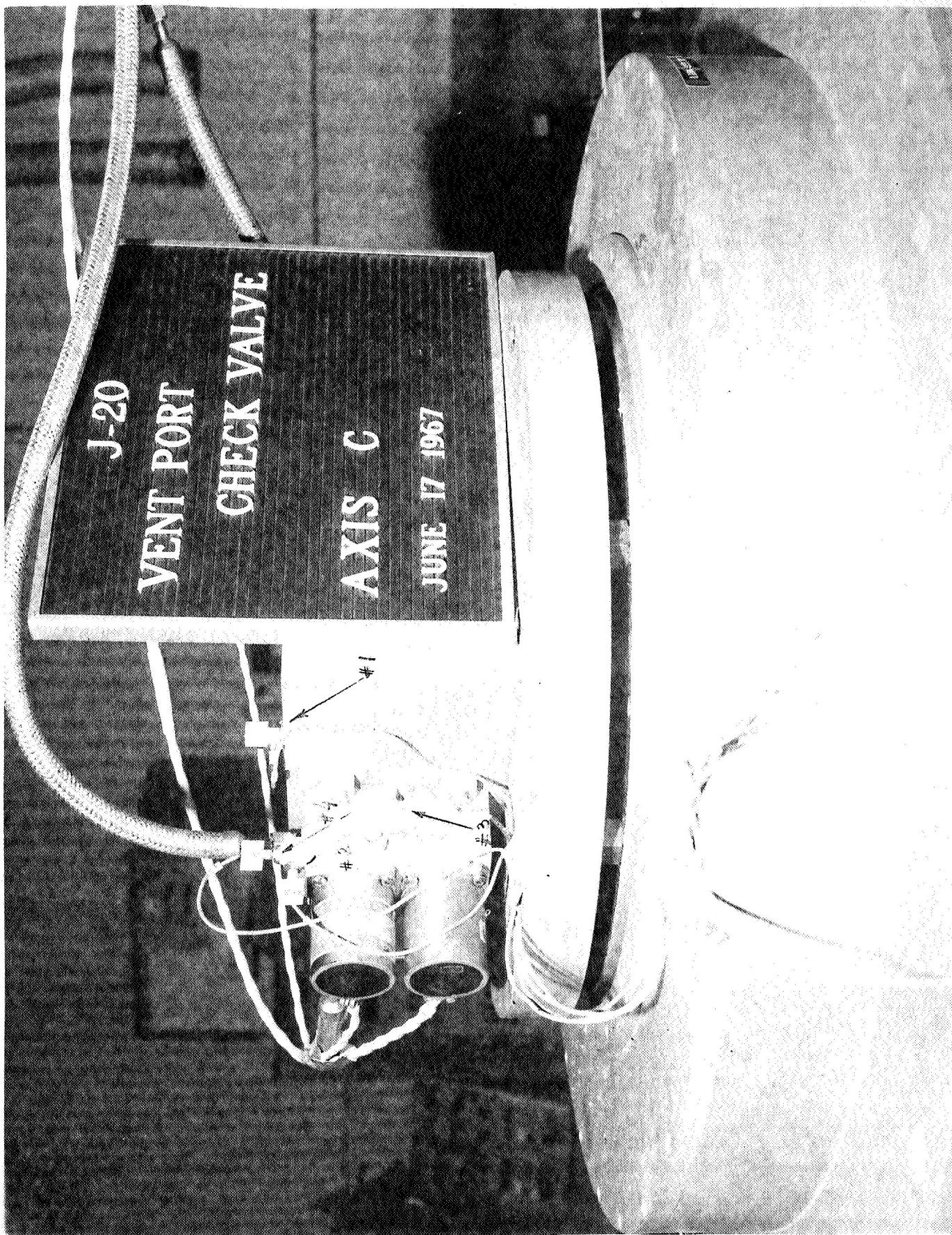
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DOUGLAS AIRCRAFT COMPANY, INC.

PAGE B 5

MODEL DSV-4B

REPORT NO. R 6033-1



Acceleration Ignition, G Axis